



An Indian Step to Space-Based Astronomical Missions



SUNIL CHANDRA
CSR, NWU
SOUTH AFRICA

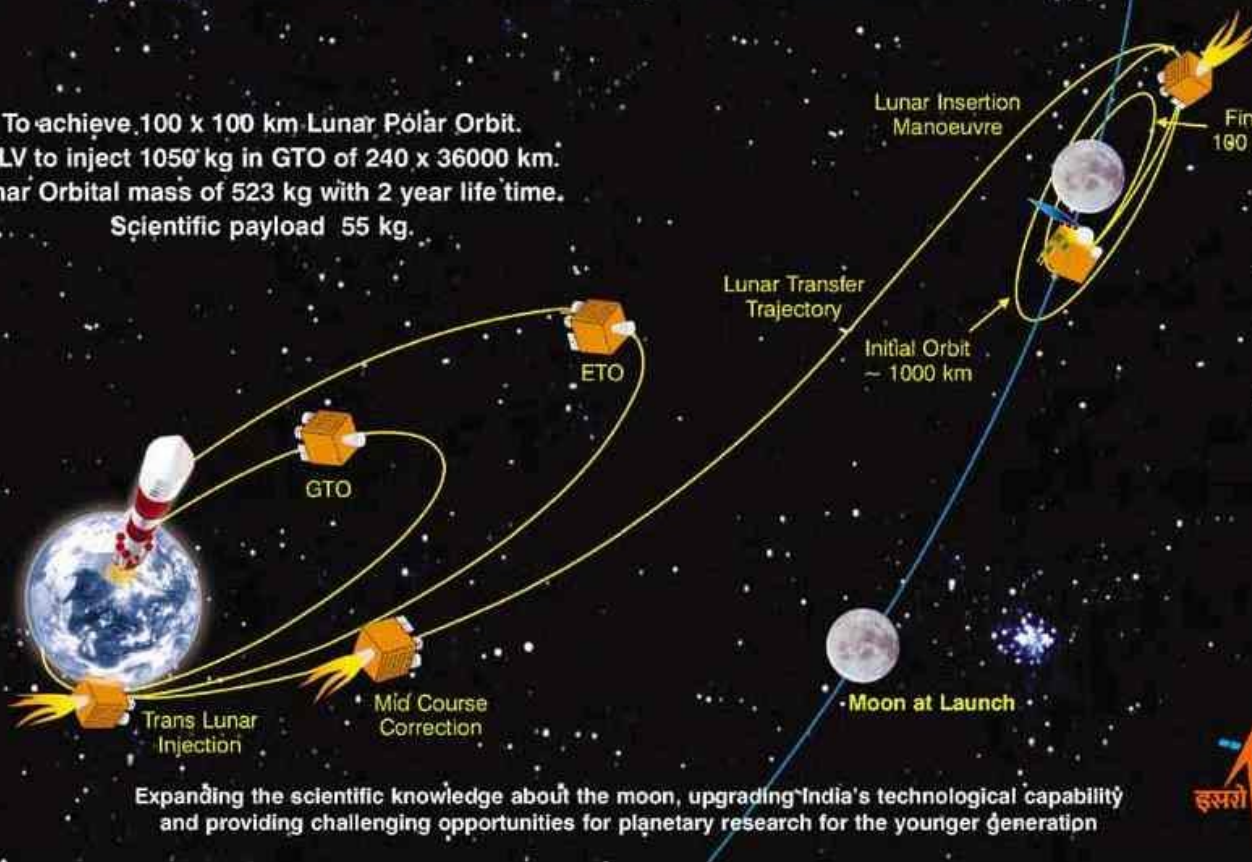
Collaborators:

M. Boettcher, K. P. Singh, S. Bhattacharyya,
G. C. Dewangan + AstroSat SXT Team

Total 97 satellite probes, till April 12, 2018

INDIA'S FIRST MISSION TO MOON CHANDRAYAAN-1

To achieve 100 x 100 km Lunar Polar Orbit.
PSLV to inject 1050 kg in GTO of 240 x 36000 km.
Lunar Orbital mass of 523 kg with 2 year life time.
Scientific payload 55 kg.



Expanding the scientific knowledge about the moon, upgrading India's technological capability and providing challenging opportunities for planetary research for the younger generation

Launched on 22 October 2008

Total 10 month 6 days life:

Discovery of water on moon...

C. M. Pieters^{1,*}, J. N. Goswami^{2,3}, et al. 2009, Sci. 326, 568

MISSION RED PLANET

MARS ORBITER

With ISRO gearing up to launch the MOM project six months from now, here's your step-by-step guide to the agency's maiden mission to the fourth planet from the sun

LAUNCH VEHICLE	
PSLV-XL with six extended strap-on motors	
MISSION COST: 7450 crore	
LIFTOFF MASS: 1,350 kg	
DRY MASS: 500 kg	
NUMBER OF PAYLOADS: Five	
POWER SYSTEM: Solar array with three panels	
ON-BOARD PROPULSION: Bi-propellant system for Mars orbit insertion	



MARS ORBITER MISSION IS ISRO'S FIRST INTERPLANETARY MISSION TO THE PLANET

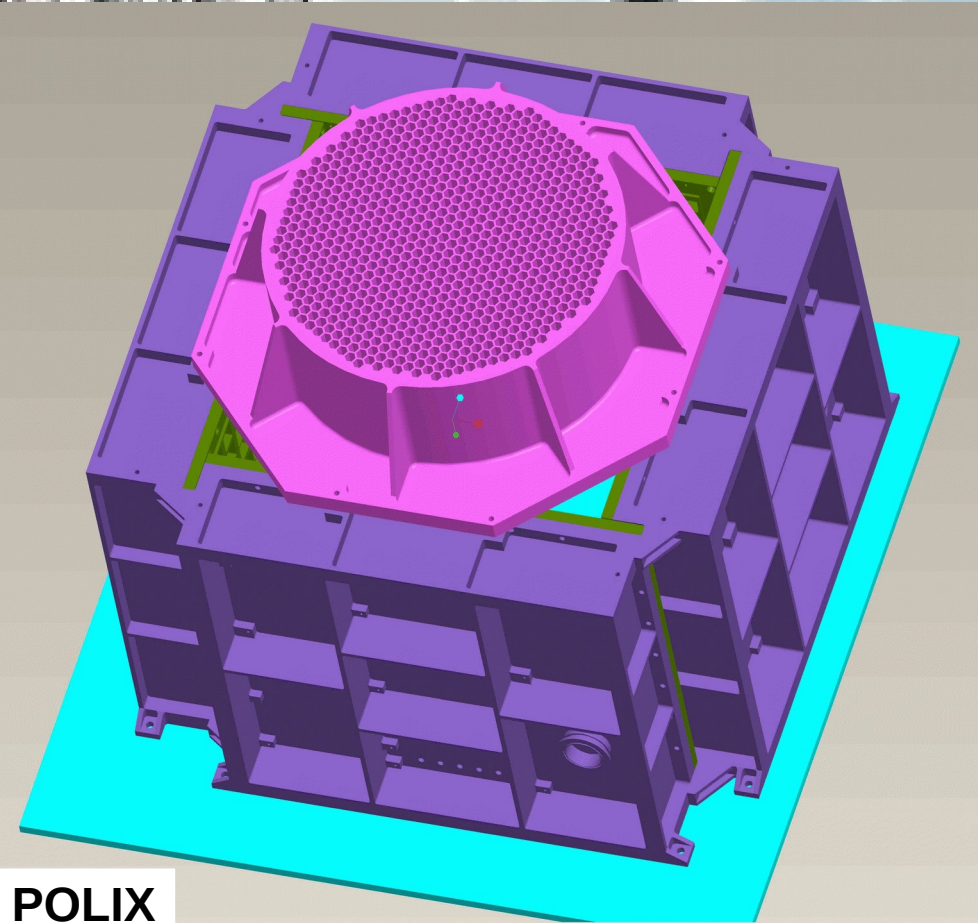


In orbit, Launched on Nov 05, 2013

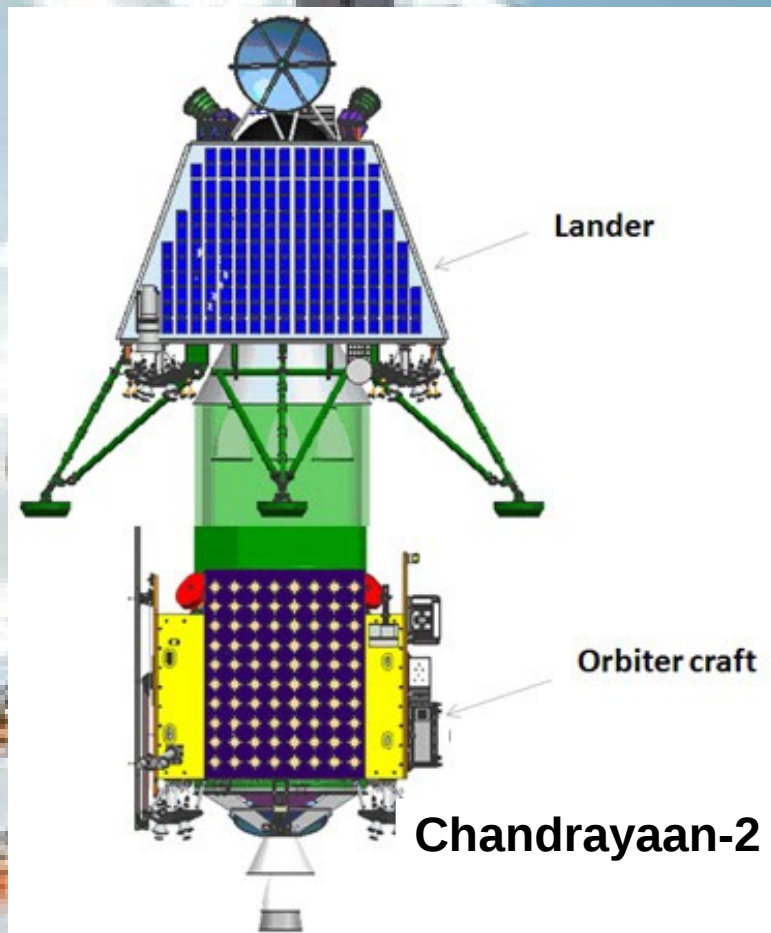
- Mars Colour Camera (MCC)
- Thermal Infrared Imaging Spectrometer (TIS)
- Methane Sensor for Mars (MSM)
- Mars Exospheric Neutral Composition Analyser (MENCA)
- Lyman Alpha Photometer (LAP)

Bhardwaj A., et al., GRL, 43, 1862, 2016 ++

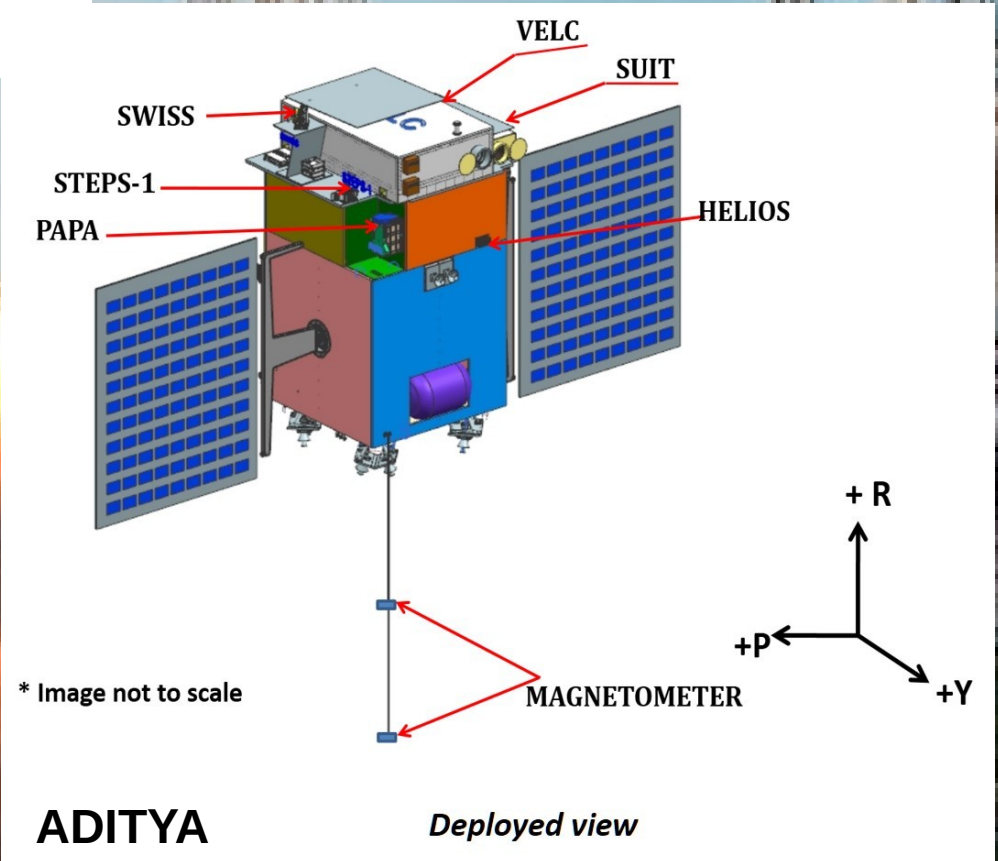
Probes planned for launches in near future...



POLIX



Chandrayaan-2



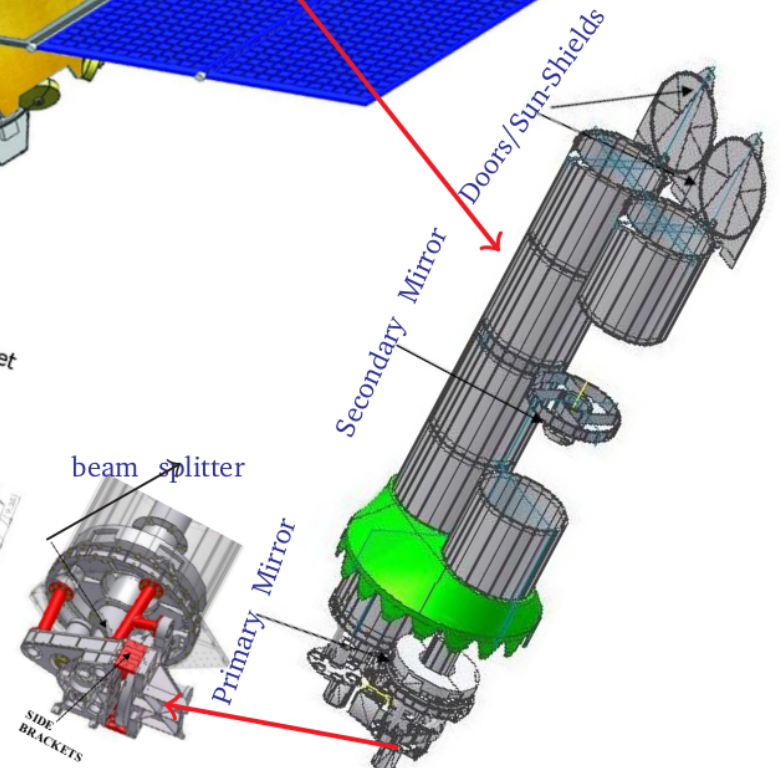
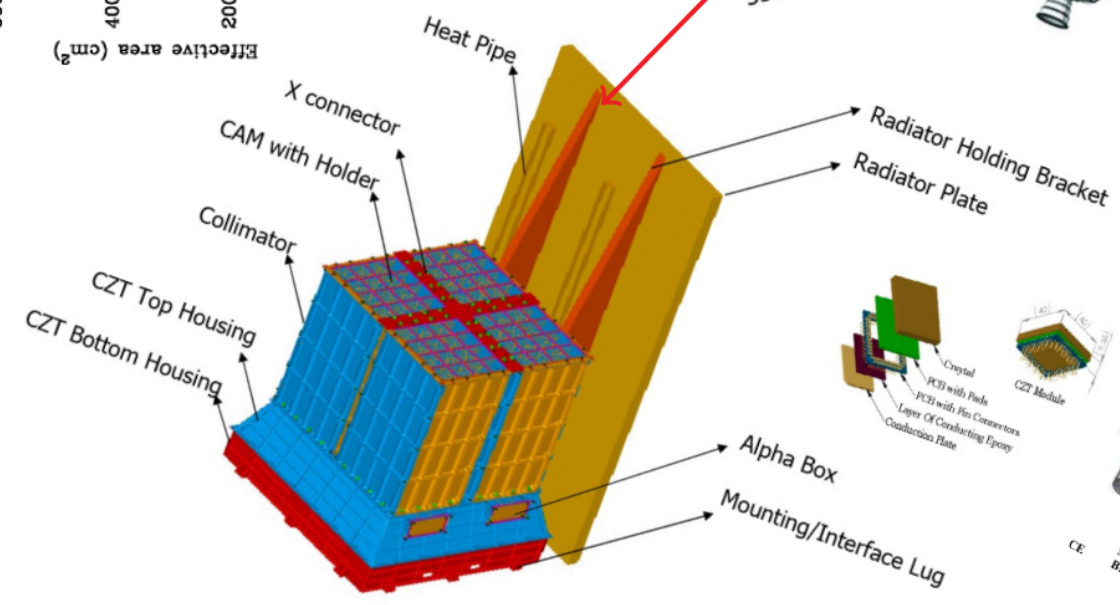
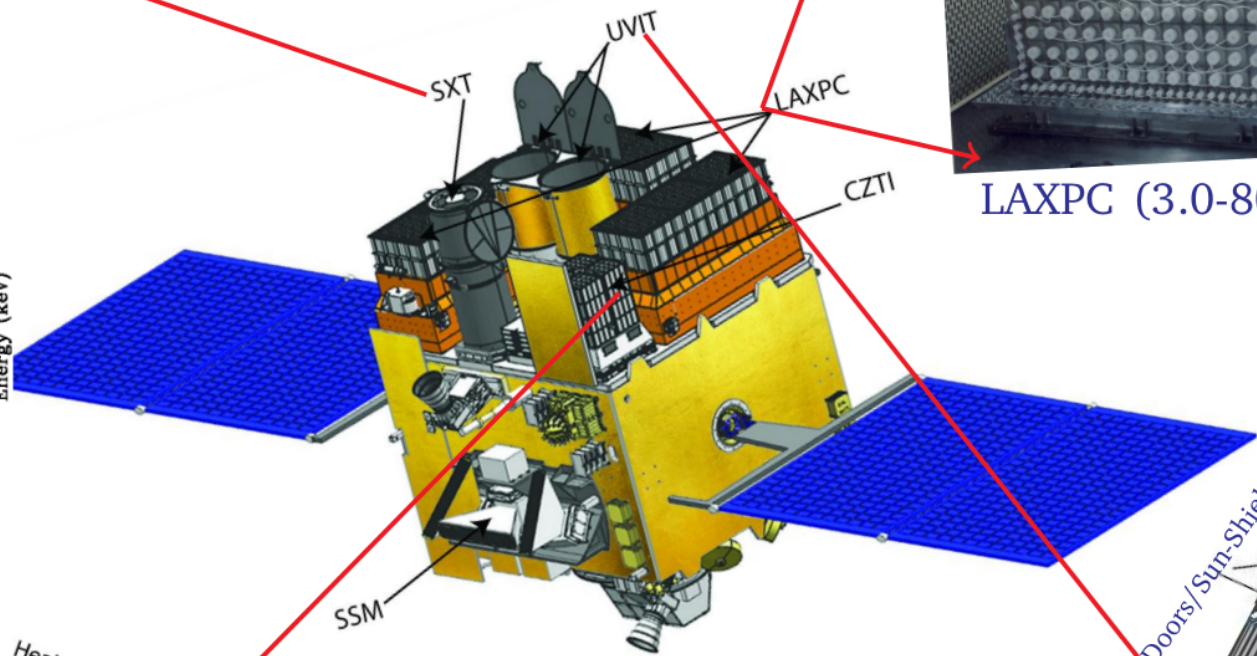
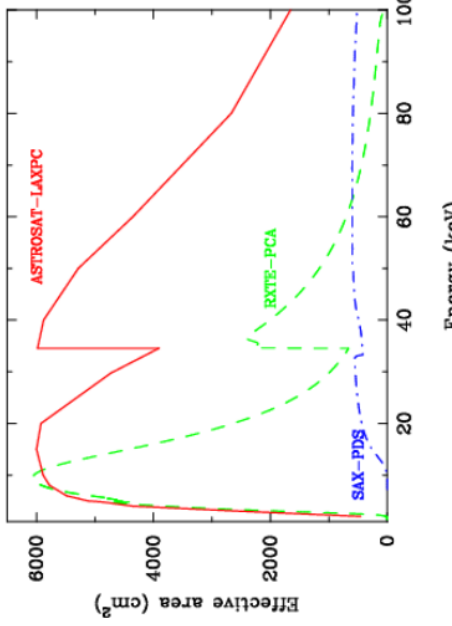
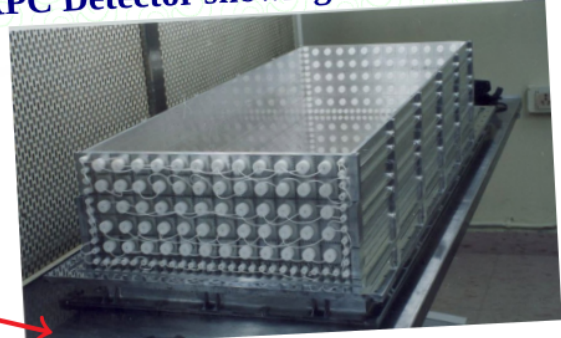
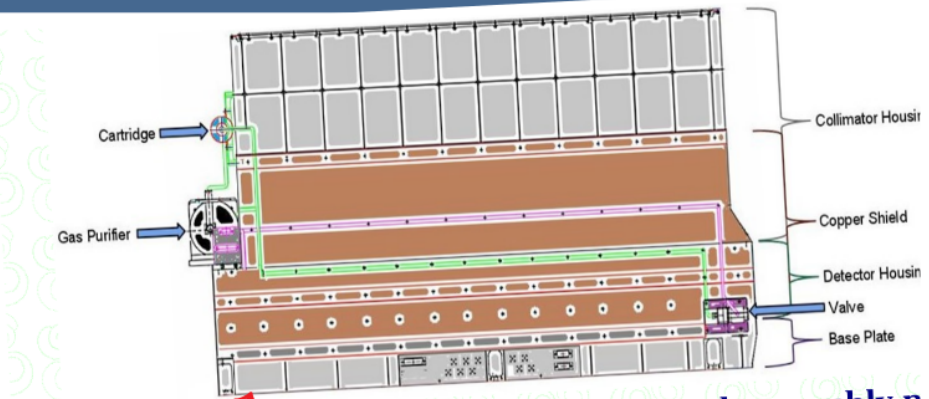
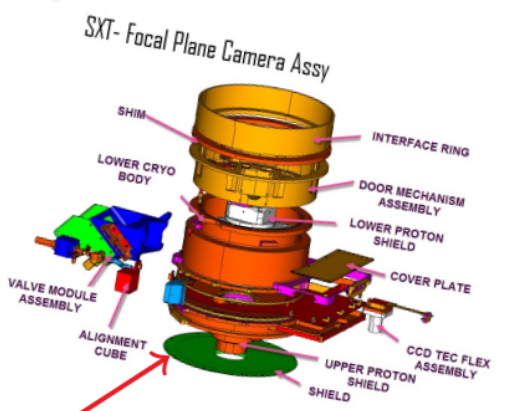
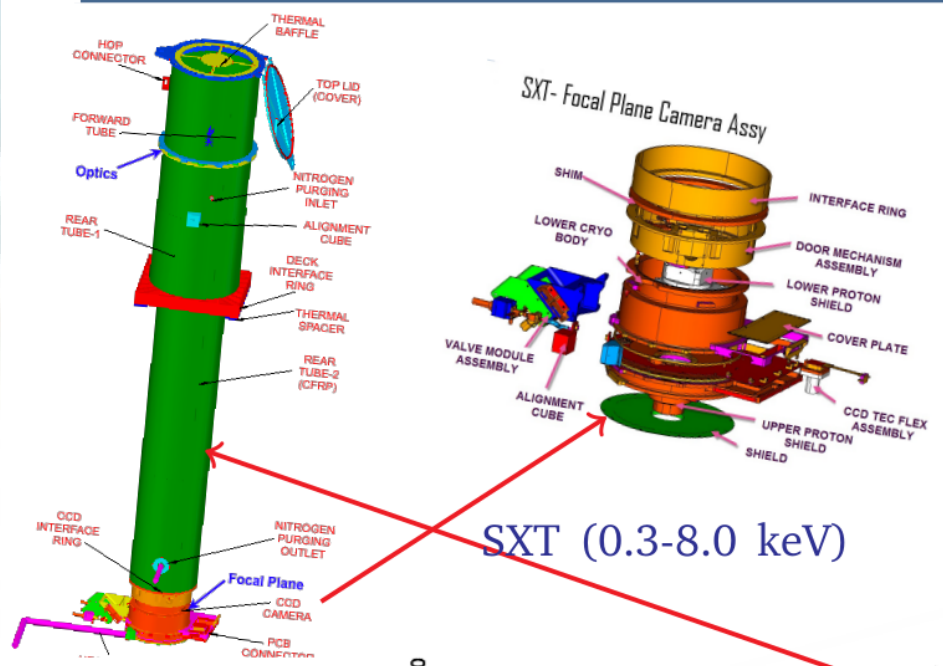
ADITYA

Deployed view



Launched on 28 September 2015

AstroSat; Introduced Launched on 28 September 2015



CZTI (10.0-150.0 keV)

UVIT (VIS/NUV/FUV Bands; 130 nm - 530 nm)

General information

Launched into a nearly circular orbit with Altitude: 640 - 650 km;
Inclination : 6 deg.

- **Orbital period** : ~98 minutes;
- **Eclipse period** : 35 minutes; **Sunlit period** : 62 minutes
- **Orientation** by 4 reaction wheels and 3 magnetic torquers (capacity: 60 A m²) + inputs from 3 dual gimbal gyros, 2 star sensors and 2 magnetometers.
- **Pointing accuracy** of ~1 arcsec with star sensors.
- **Drift rate** is expected to be 0.2 arcsec/s.
- **Maximum slew rate** will be 0.6°/s.
- **Solid-state recorder** with 200 Gb storage (4 orbits).
- **Operational life** of > 5 years

Cadmium Zinc Telluride Imager (

Area: 976 cm²

Pixels: 16384 (64 modules of 256 pixels each)
Pixel size 2.46 mm X 2.46 mm (5 mm thick)

Imaging method Coded Aperture Mask (CAM)

Field of View (10-100 keV):

4.6° X 4.6° FWHM (primary FOV)

11.8° X 11.8° FWZM (incl. illumination leakag

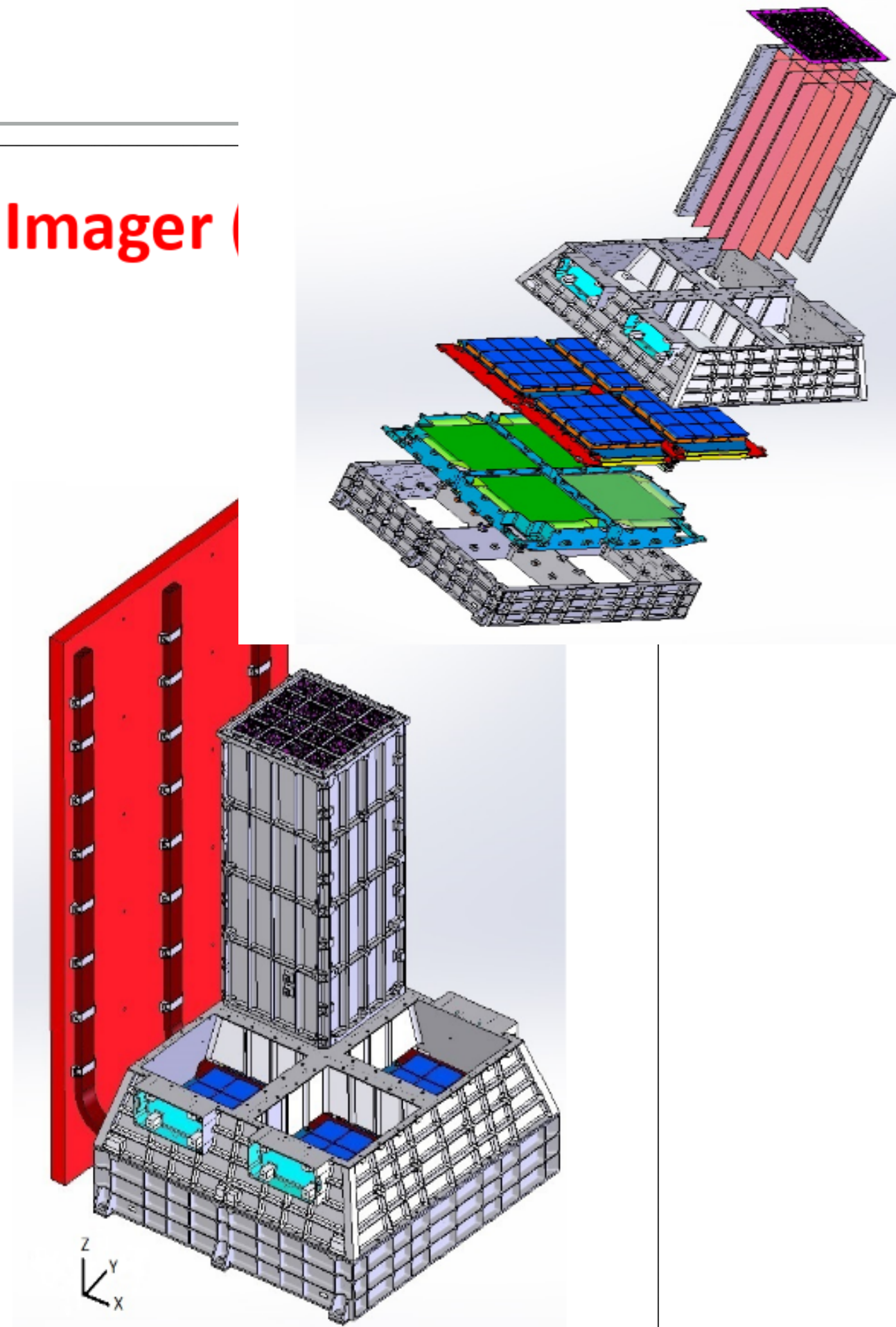
Angular resolution ~ 8 arcmin

Energy resolution ~ 8% @ 100 keV

Energy range: 10 – 100 keV

Up to 1 MeV (Photometric); no imaging above 100 keV

Sensitivity: 0.5 mCrab (5 sigma; 10⁴ s)

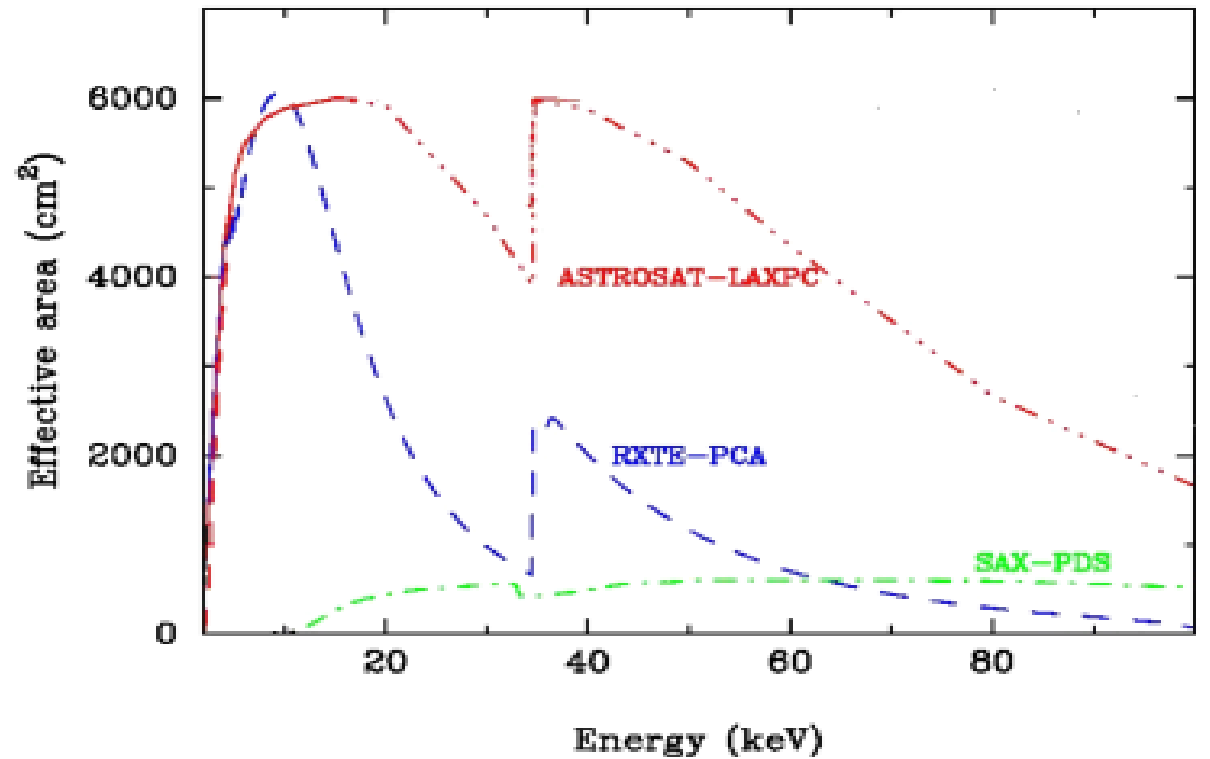


Payloads, LAXPC



5 layers, 60 anode cells; veto layer on 3 sides

<http://lanl.arxiv.org/pdf/1702.08624v2>



FOV: $1^\circ \times 1^\circ$ $E \sim 3.0 - 80$ keV

High Detection Efficiencies

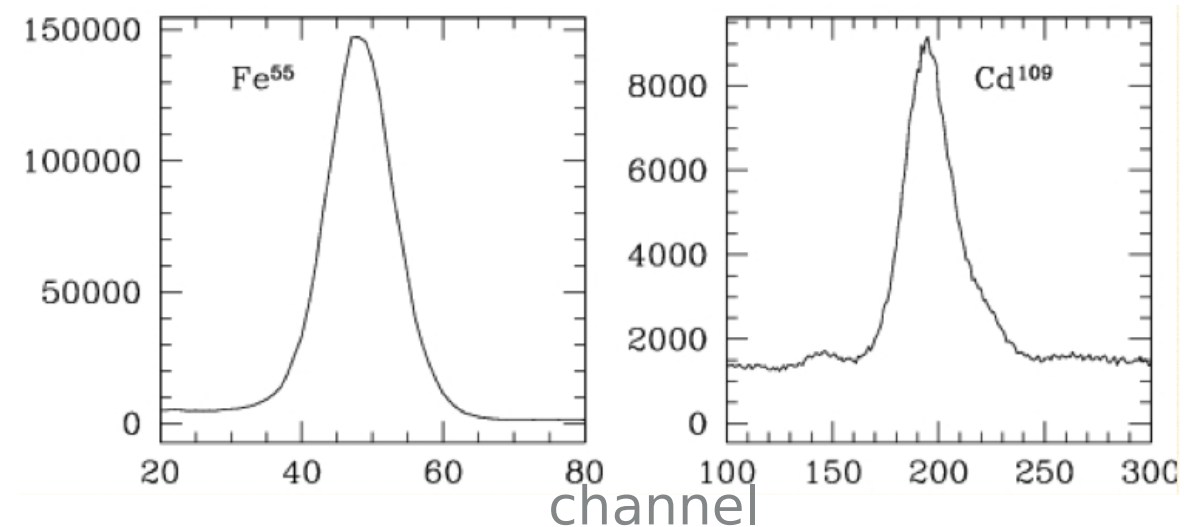
(100% $E < 15$ keV & 50 upto 80 keV)

Moderate Energy Resolution

Timing accuracy: 20 micro sec

Sensitivity : 0.1 mCrab at 5 sigma for 10^4_s

Small Internal Background



Payloads, Ultra-Violet Telescope (UVIT)



Table 1 Specifications of Ultra Violet Imaging Telescope

	FUV	NUV	VIS
Detector	Intensified CMOS Photon Counting/ Integration	Intensified CMOS Photon Counting/ Integration	Intensified CMOS Photon Counting/ Integration
CMOS chip	Fillfactory/ Cypress STAR250, 512x512, 25µm pixels		
Telescope Optics	Ritchey-Chertian 2 mirror System	Ritchey-Chertian 2 mirror System	Ritchey-Chertian 2 mirror System
Bandwidth	130-180 nm	200-300 nm	320-550nm
Geometric Area (cm ²)	~880	~880	~880
Effective Area (cm ²)	> ~15 at peak	> ~50 at peak	> ~50 at peak
Field of View	~28'	~28'	~28'
Spectral Resolution	<1000 Å (depends on Choice of Filters)	<1000 Å (depends on Choice of Filters)	<1000 Å (depends on Choice of Filters)
Spatial Resolution	<1.8 arc second	<1.8 arc second	<1.8 arc second
Time Resolution	<10 ms (for Partial field)	<10 ms (for Partial field)	<10 ms (for Partial field)
Typical Observation time per target	30 min	30 min	30 min
Sensitivity (obs.time)	>20 ¹³ Magnitude (5σ) in 200 s	-	-
Photometry Accuracy	10%		
Total Mass (Kg)	230 Kg		
Total Power (watts)	85 watts (peak 117 watts)		
Sun-avoidance angle	45 deg		

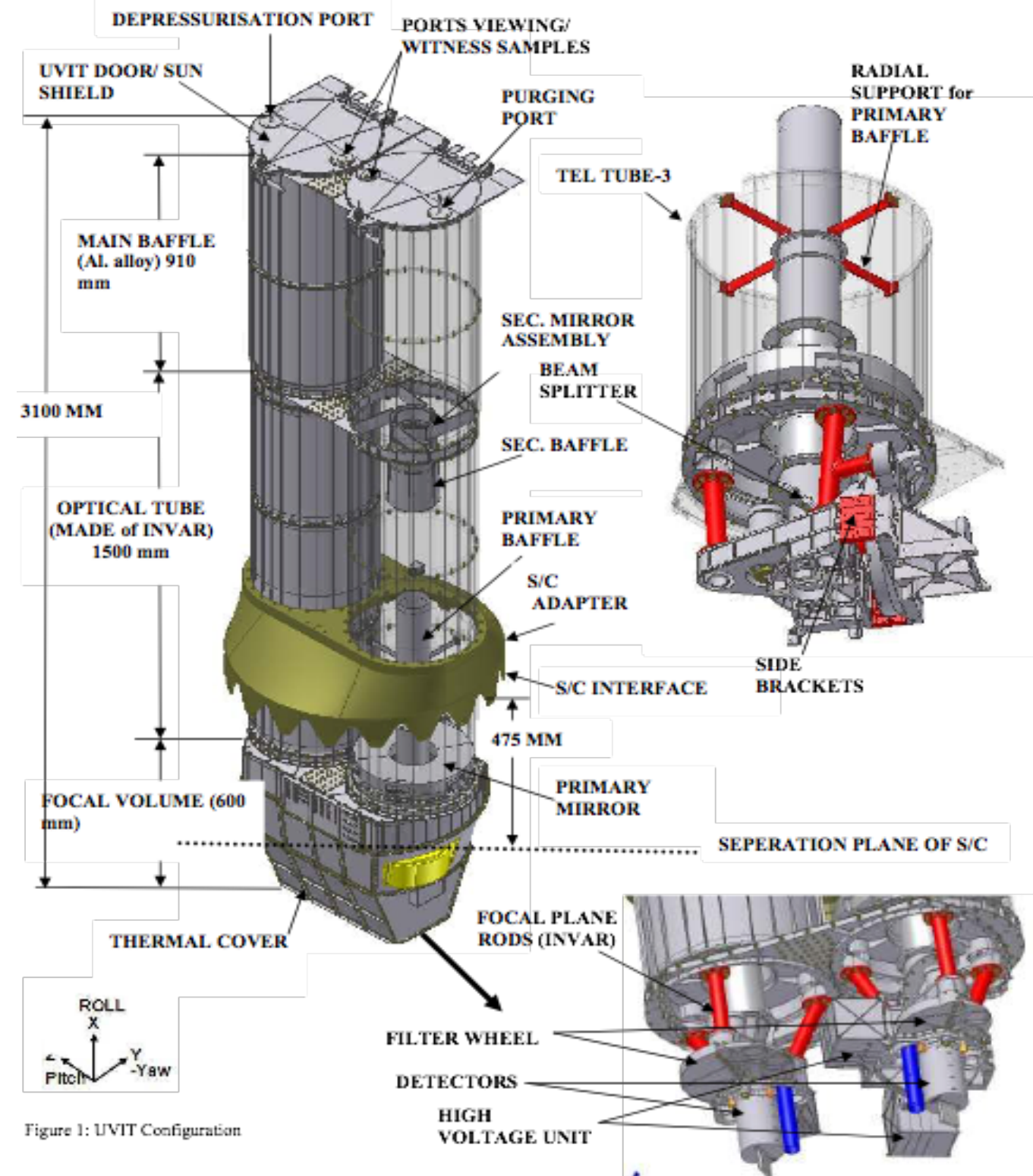
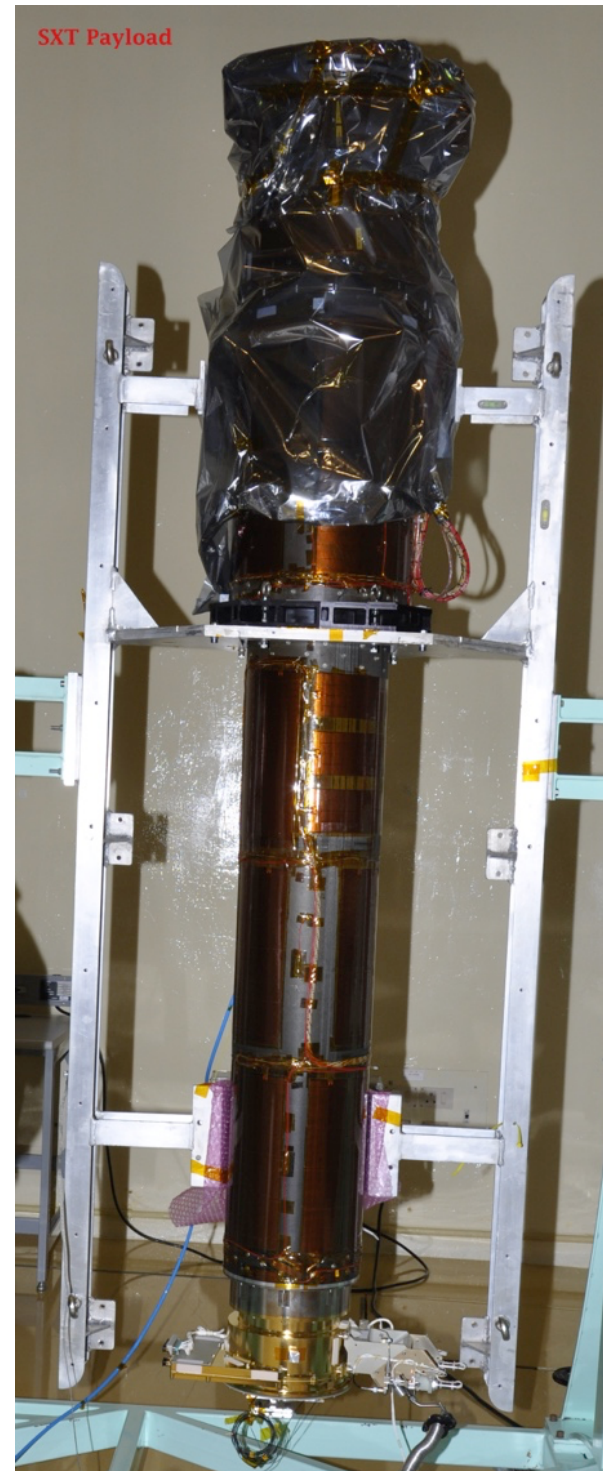
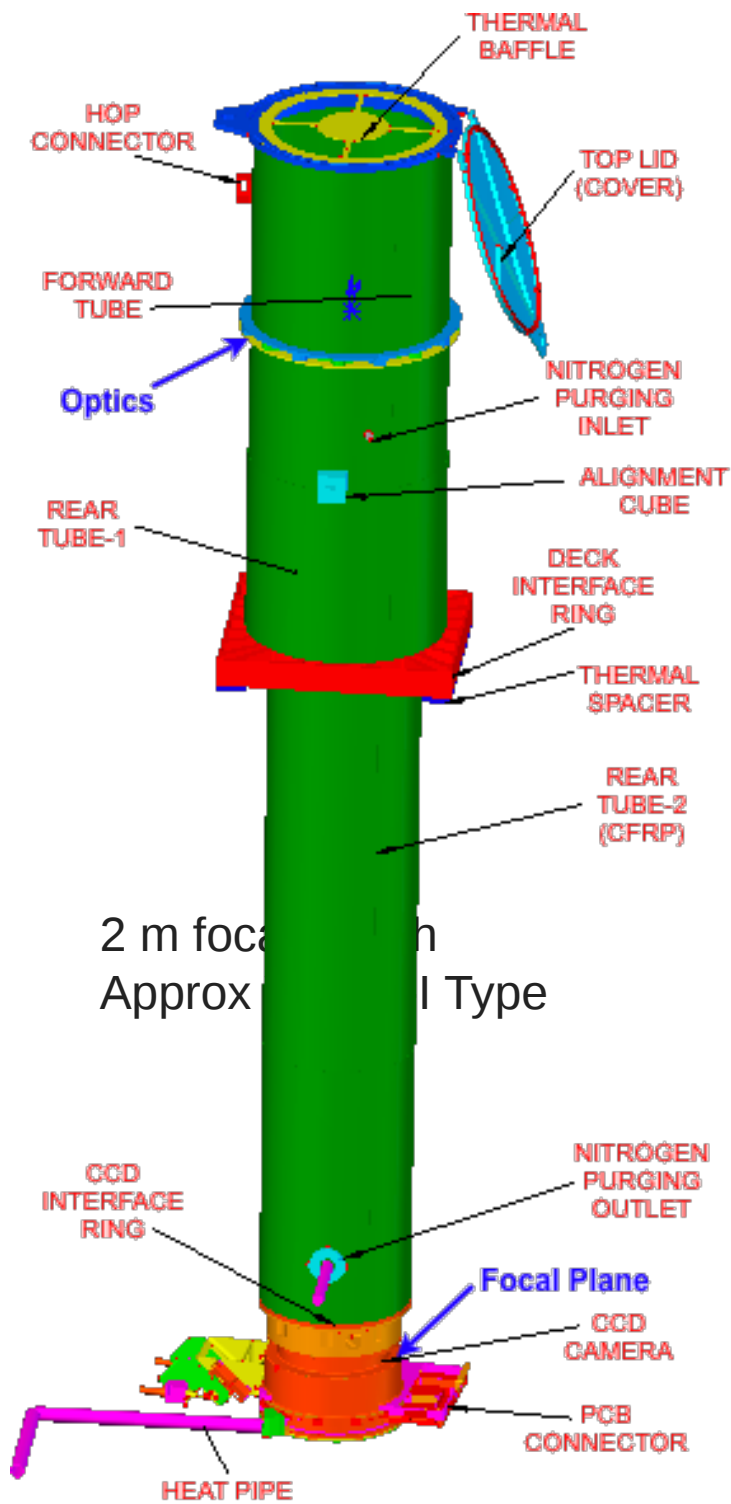


Figure 1: UVIT Configuration

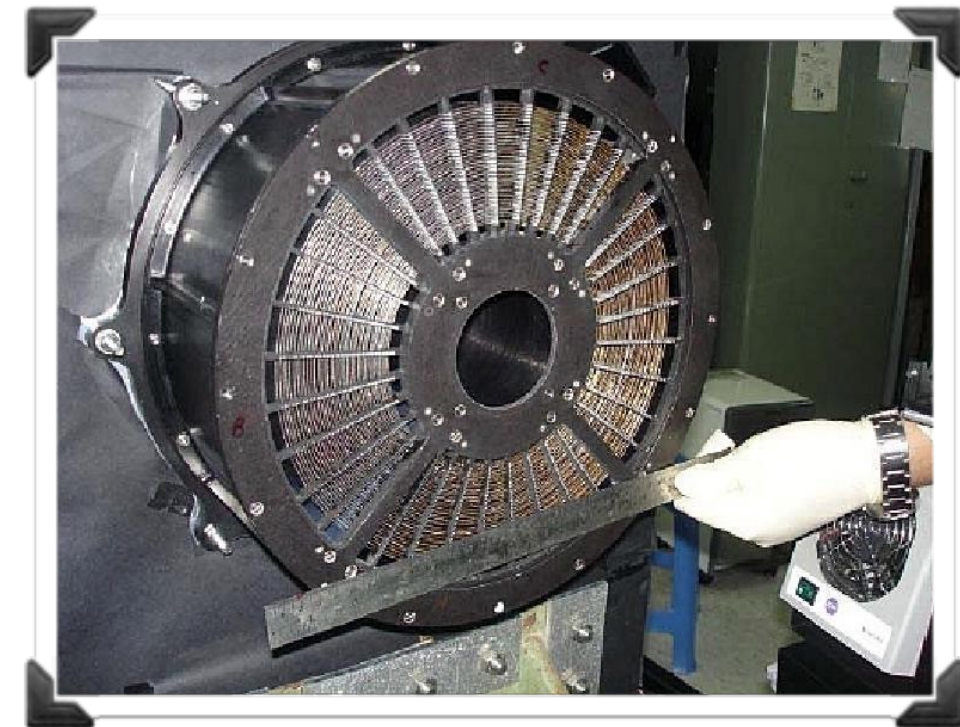
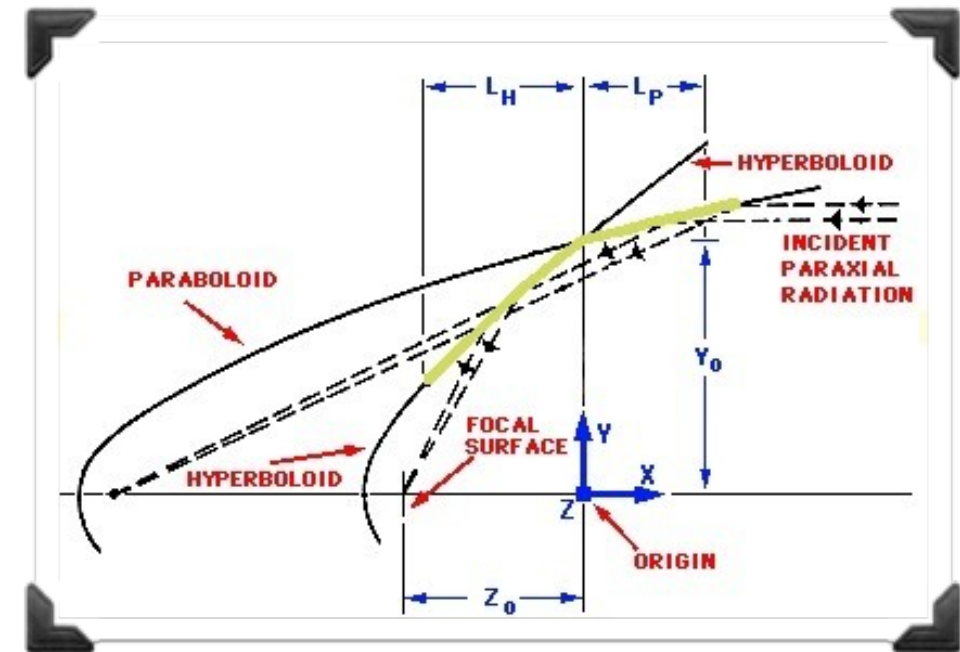
Twin 38 cm telescope

Payloads, Soft X-Rays Telescope (SXT)



0.3-8 keV
Focussing Telescope
FOV 40 arcmin
Resoln 3 arcmin
Sensitivity
~10 microCrab

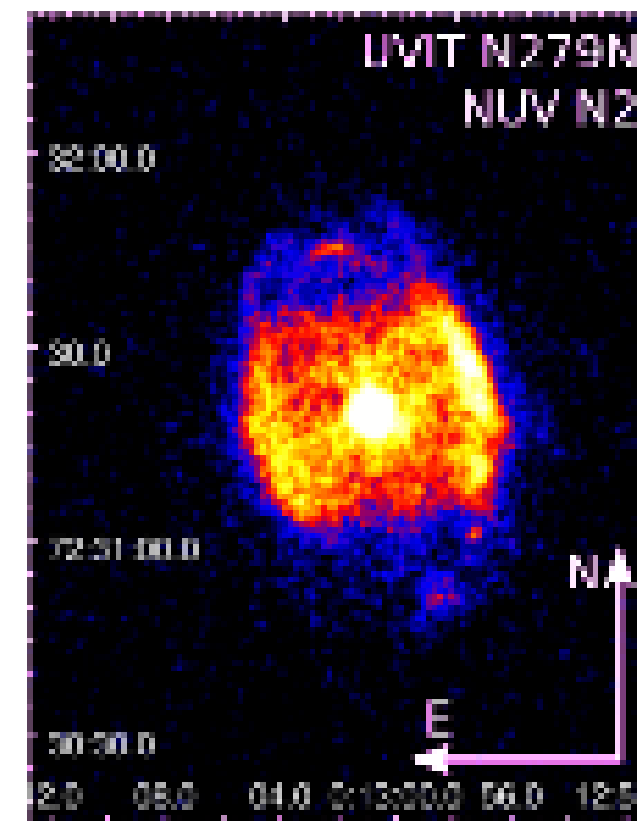
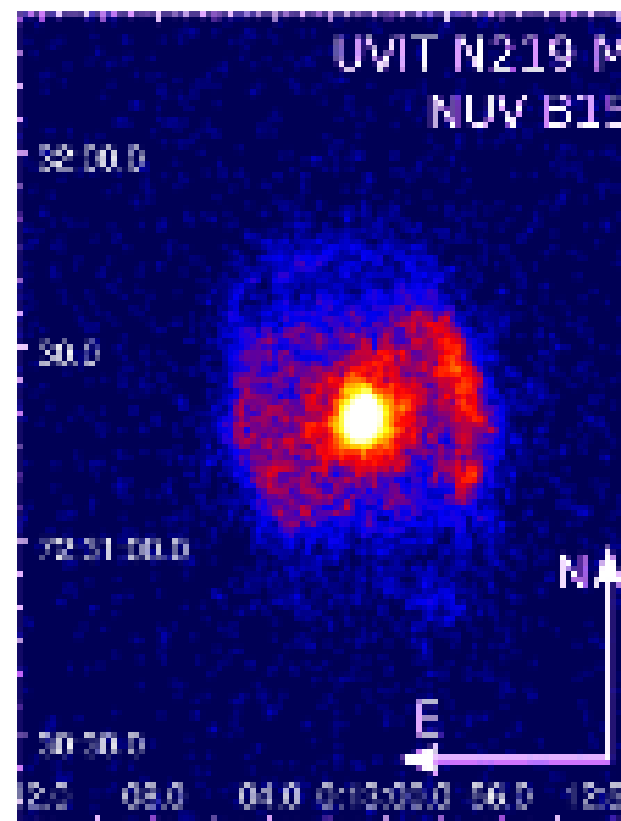
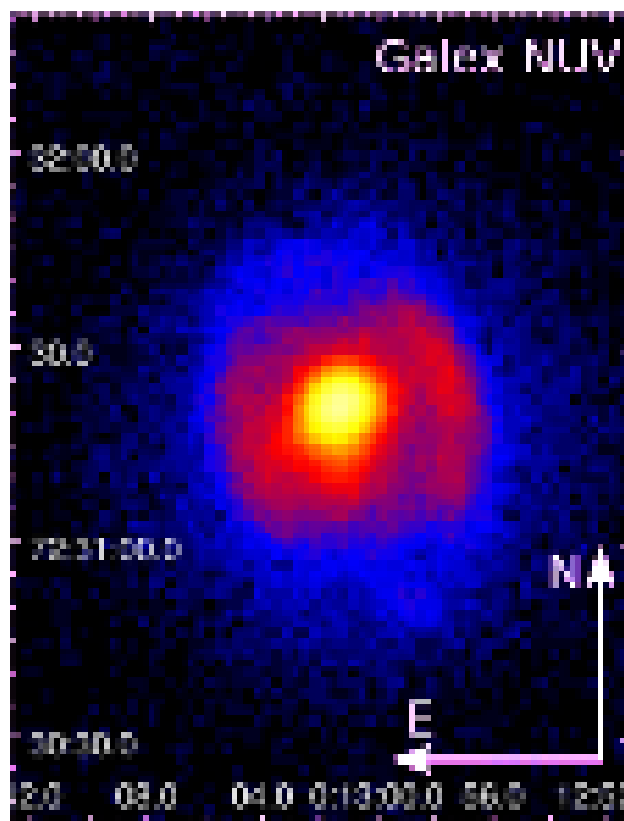
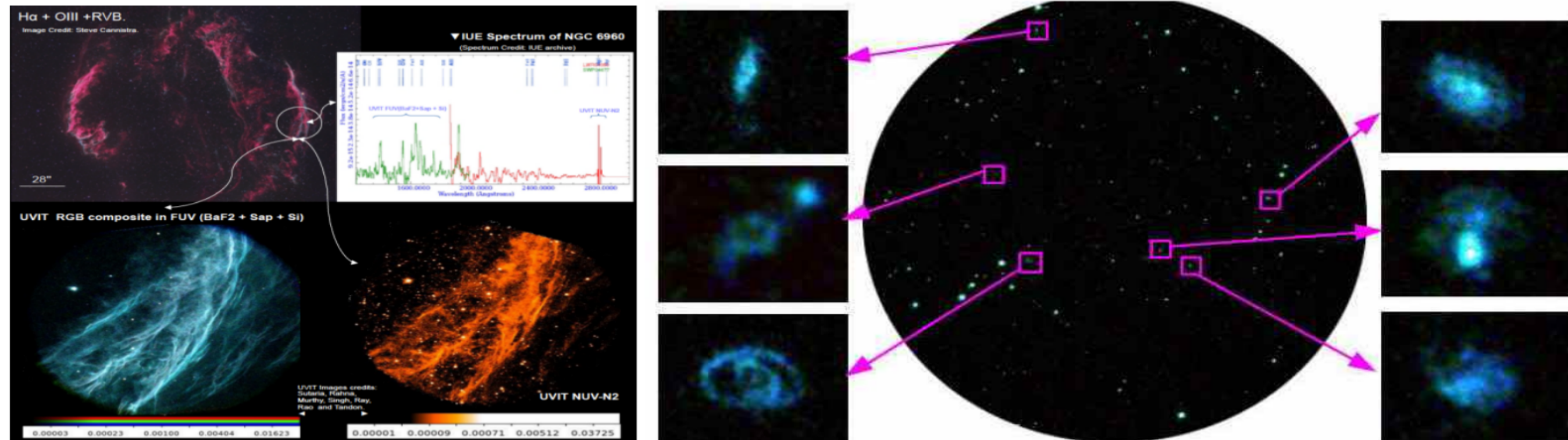
Mirrors: Nested & Segmented conical surfaces in Wolter type I geometry working at Grazing incidence. 39 nested shells



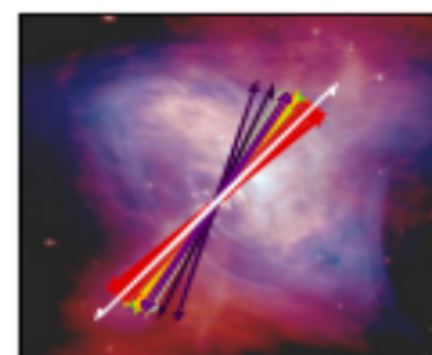
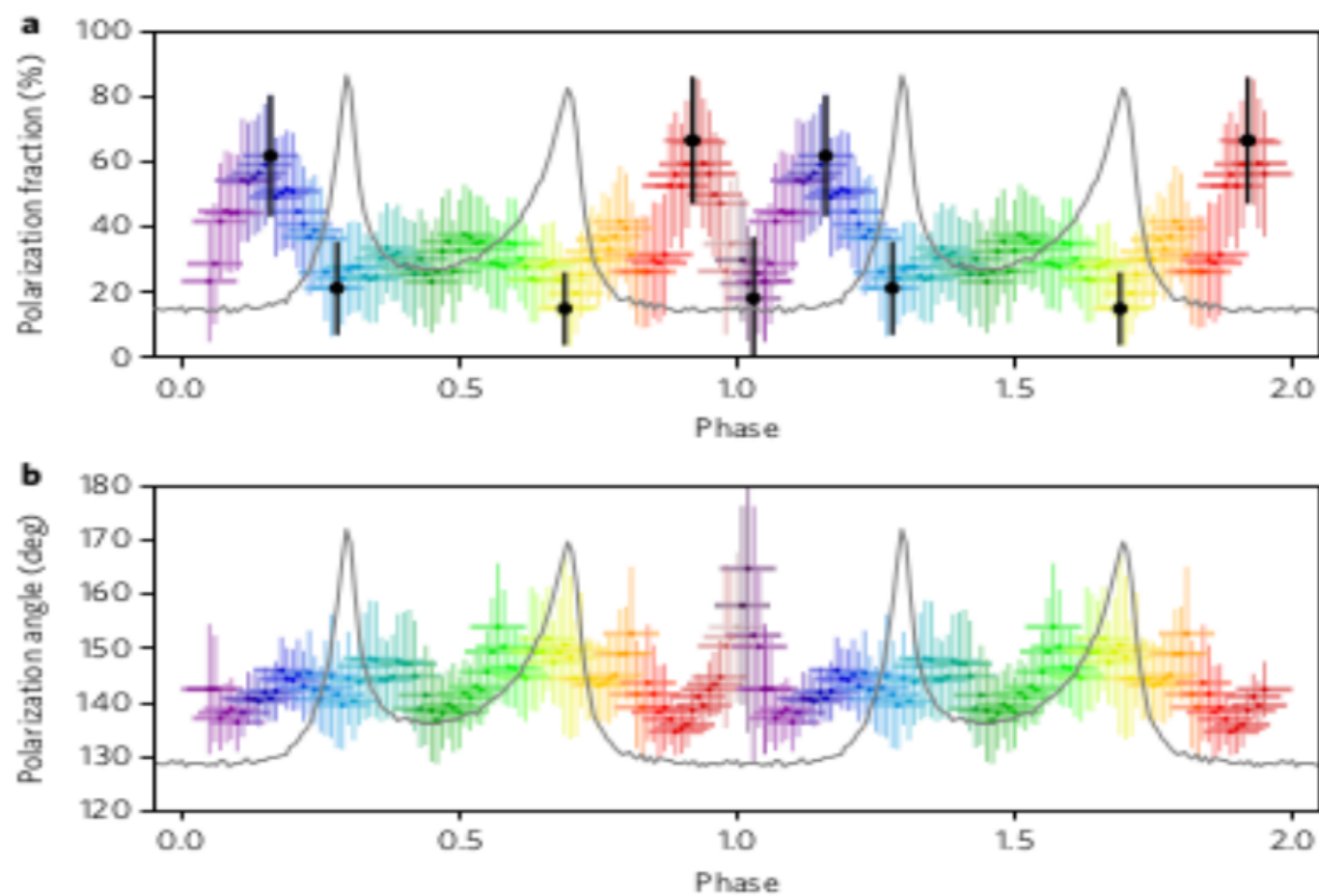
Telescope PSF	:	1.5 - 2.5 arcmin (rms)
---------------	---	------------------------

UVIT NGC 40 (Bow Tie Nebula)

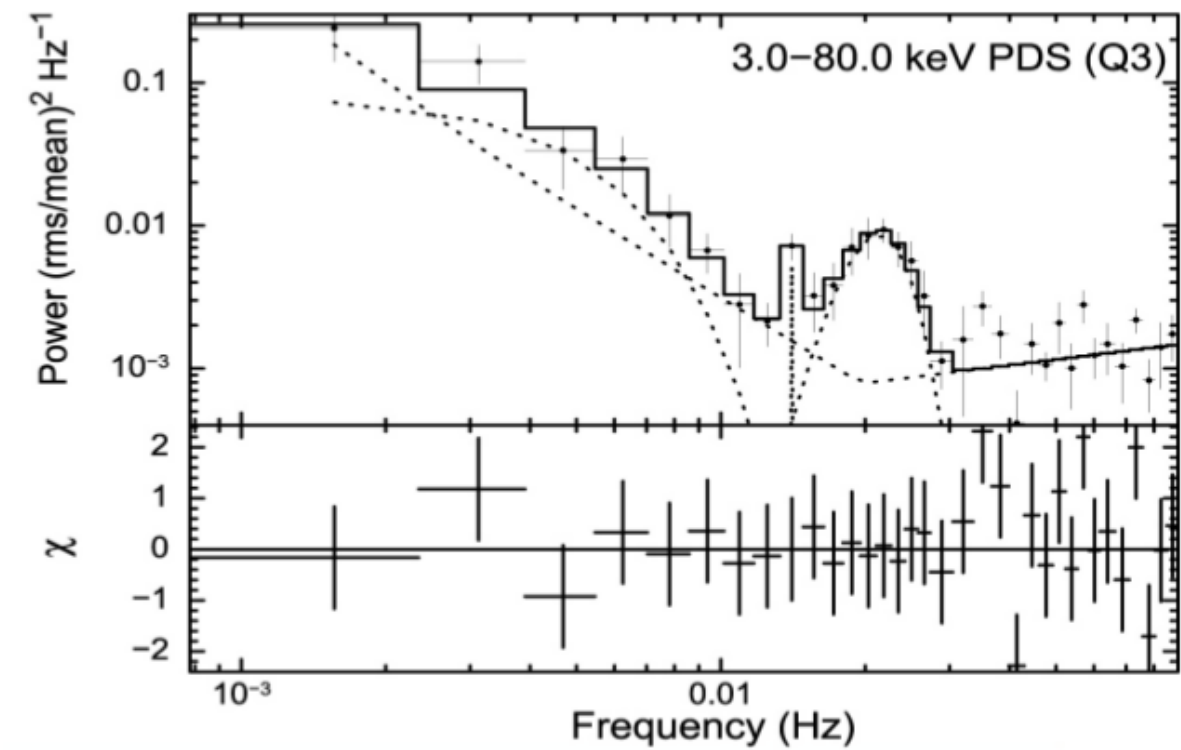
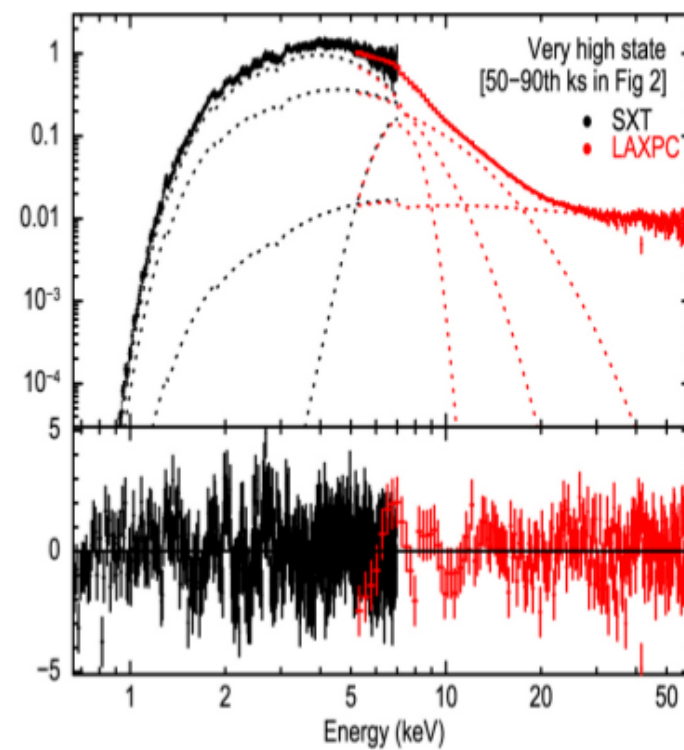
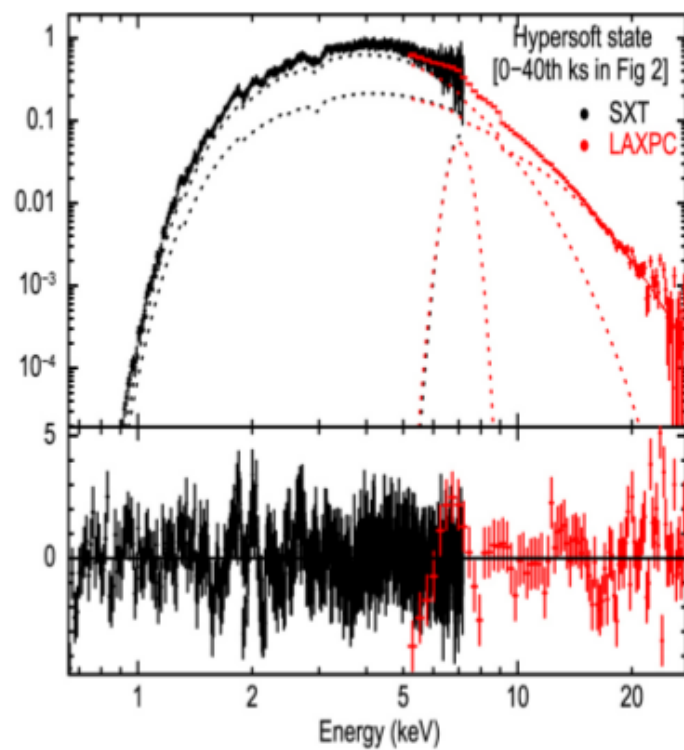
- Stunning UV Images of extended structures in our nearby universe : (Courtesy: UVIT Team and AstroSat Picture of the month; e.g. <https://www.isro.gov.in/update/02-jul-2018/>)



-
- Hard-Xray (>100 keV) polarization from Crab, GRBs and Cygnus X-1 (Example Phase-resolved X-ray polarimetry of the Crab pulsar, Vadawale et al., Nature Astronomy 2, 50, 2018)



Spectral and Timing Studies of XRBs (Example Cyg X-3, Pahari et al 2018a, 2018b)



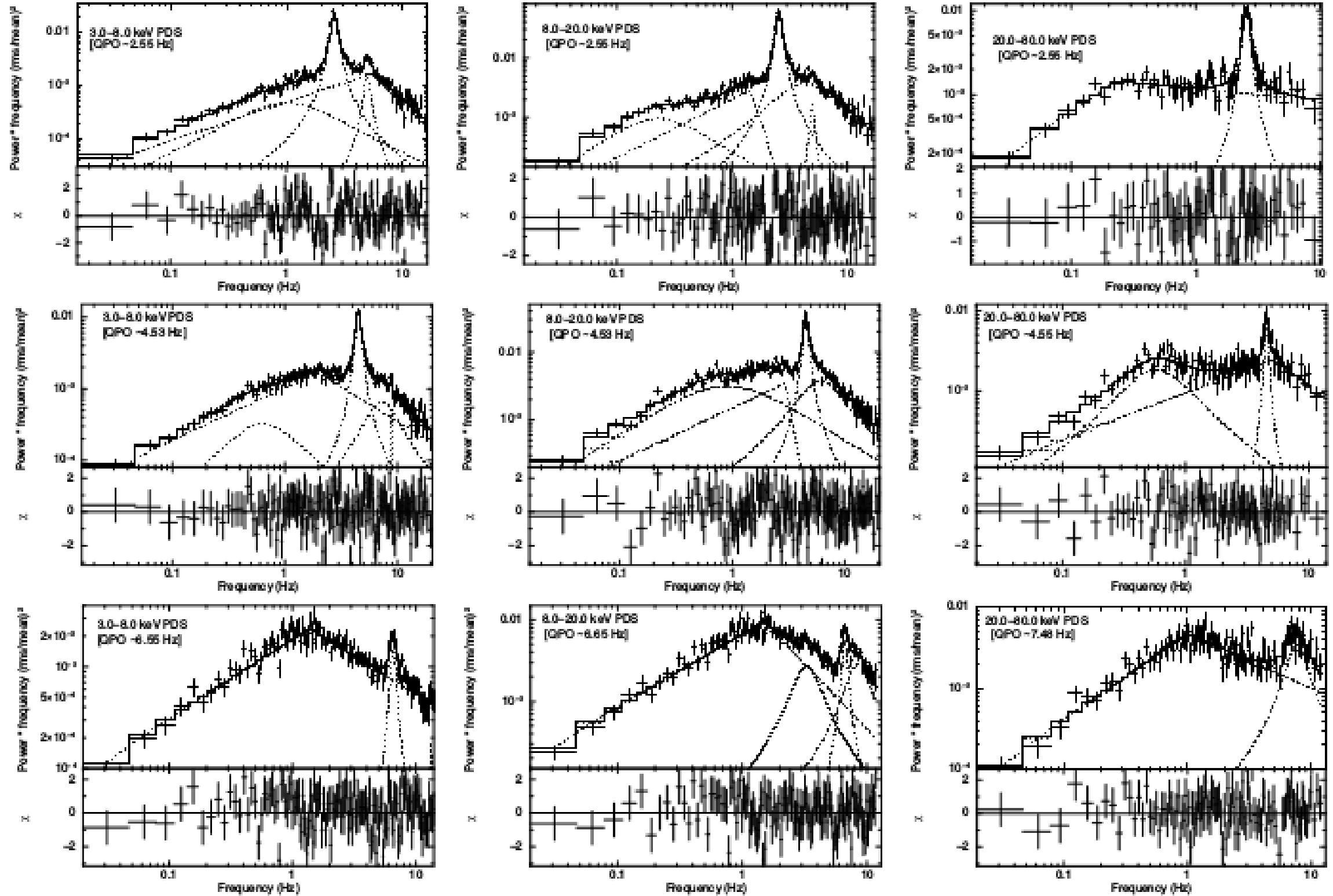
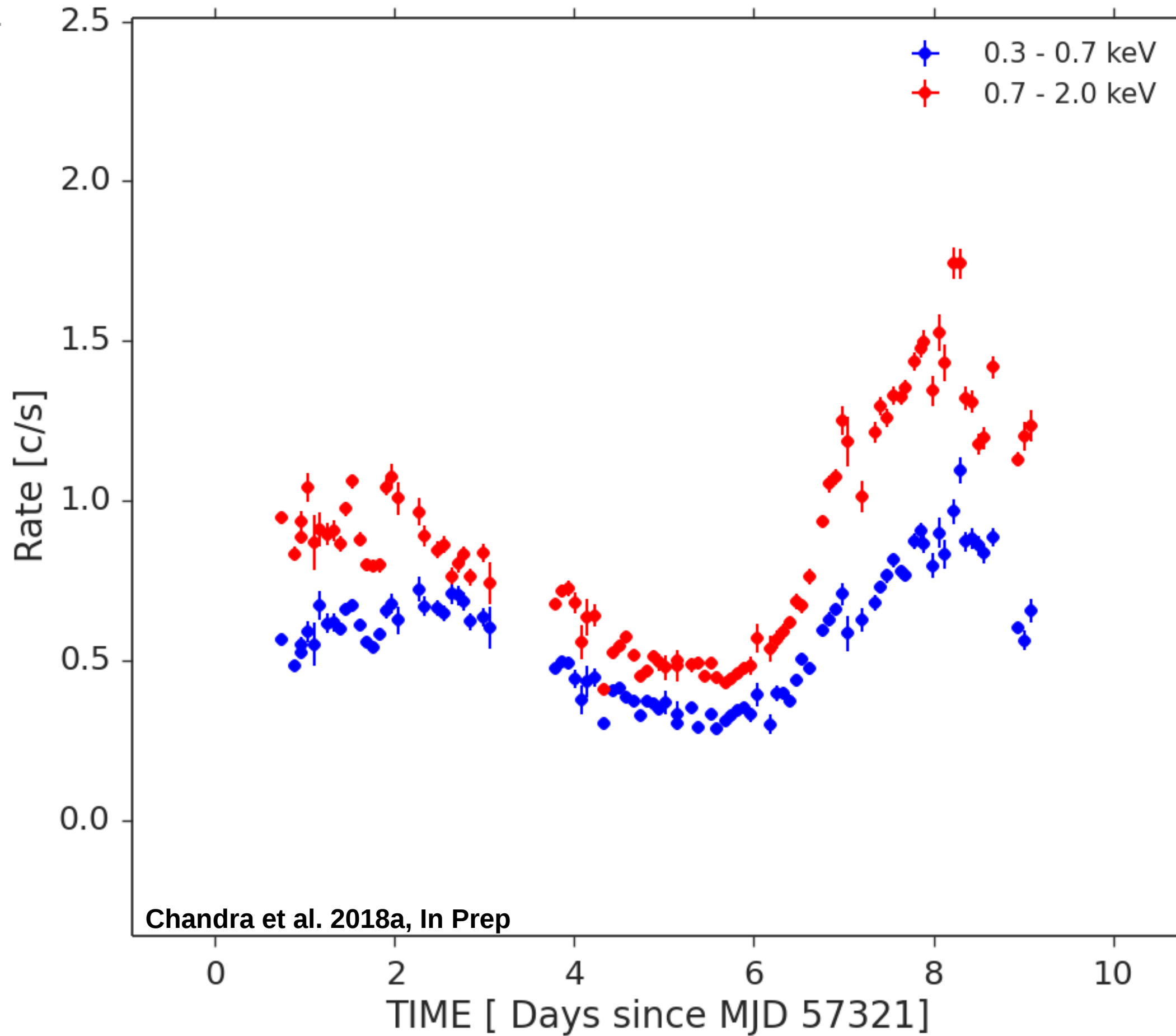
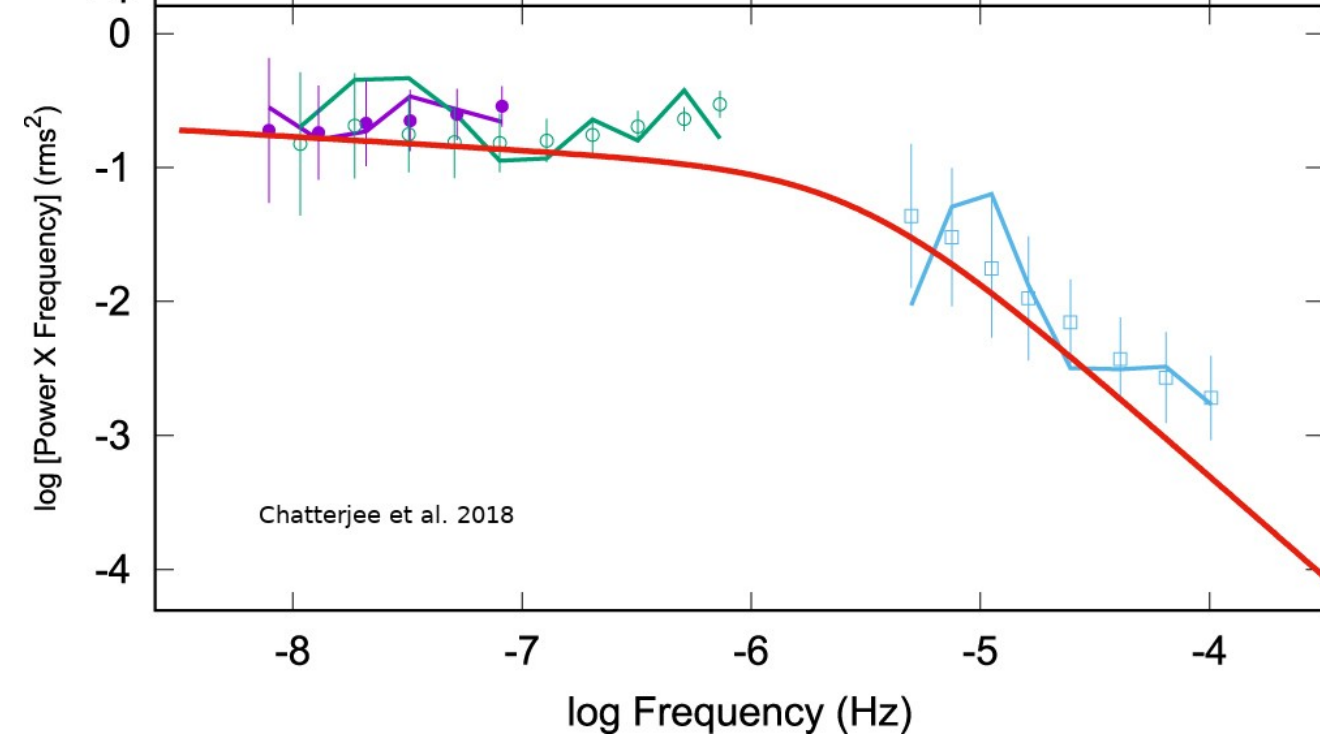
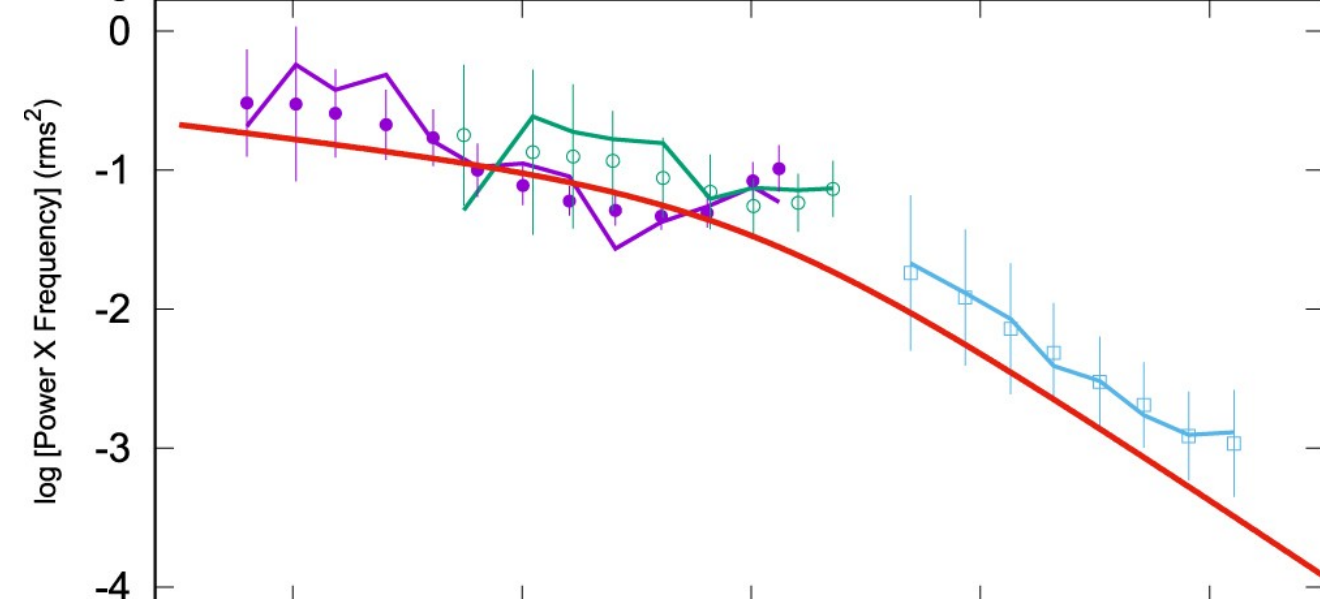
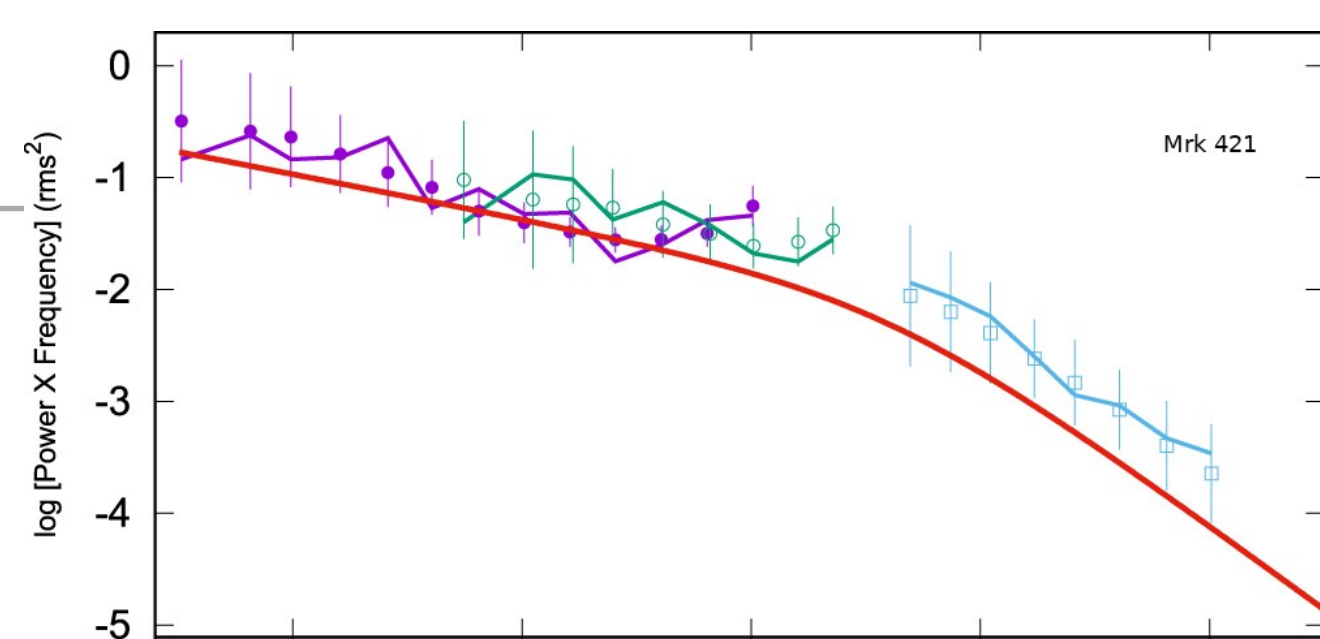
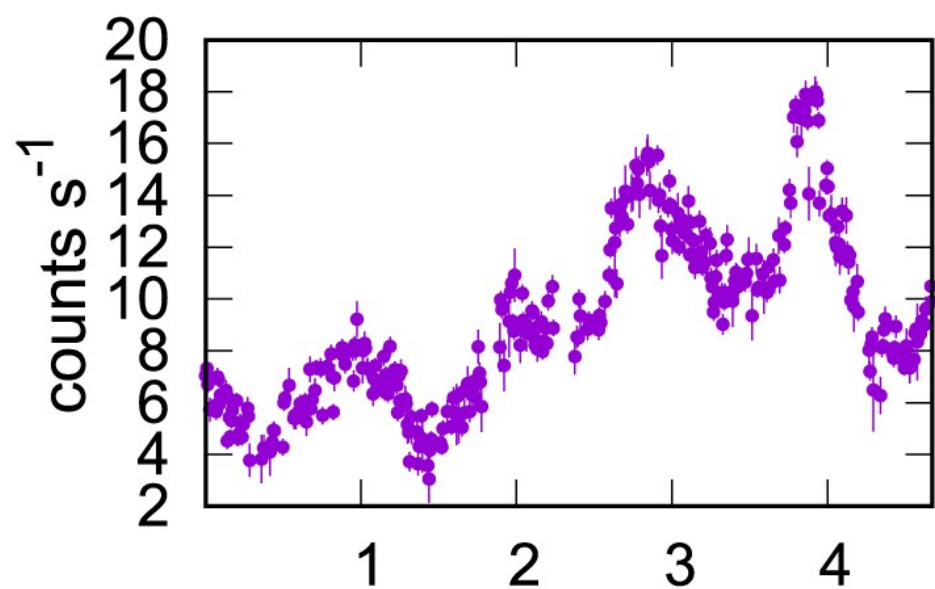
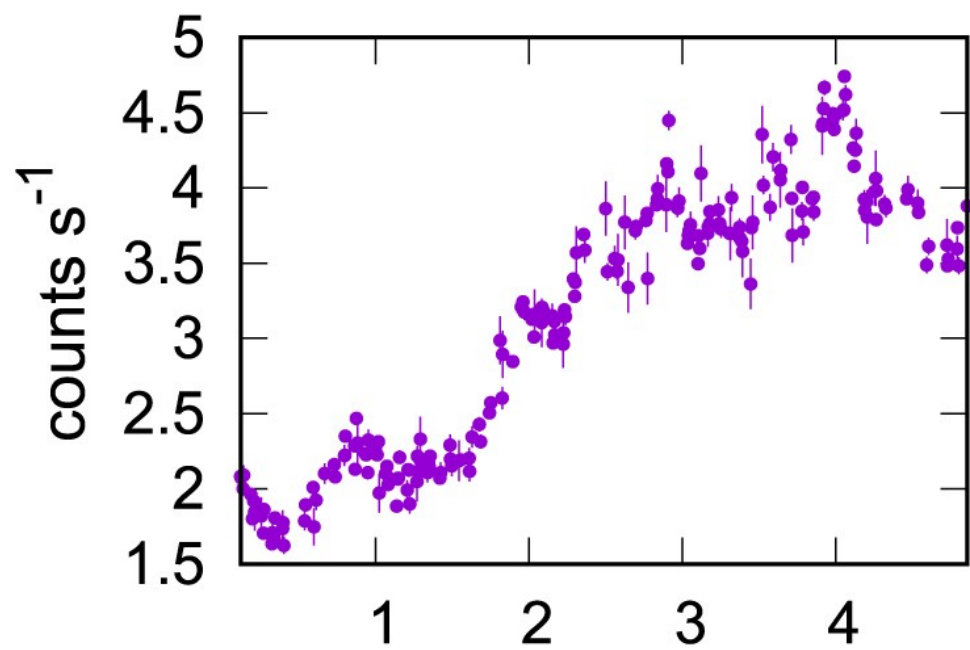
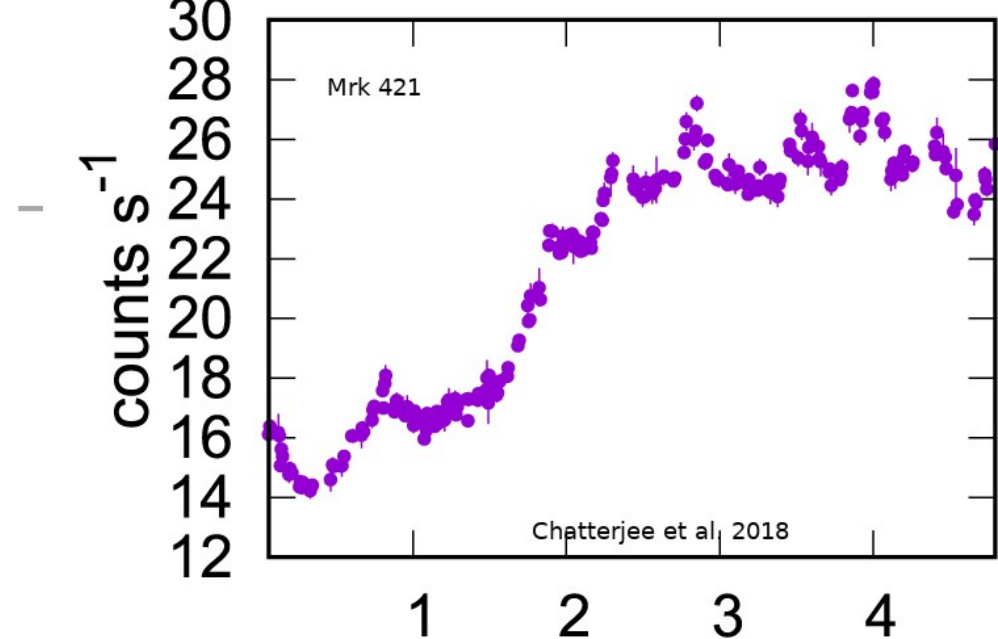
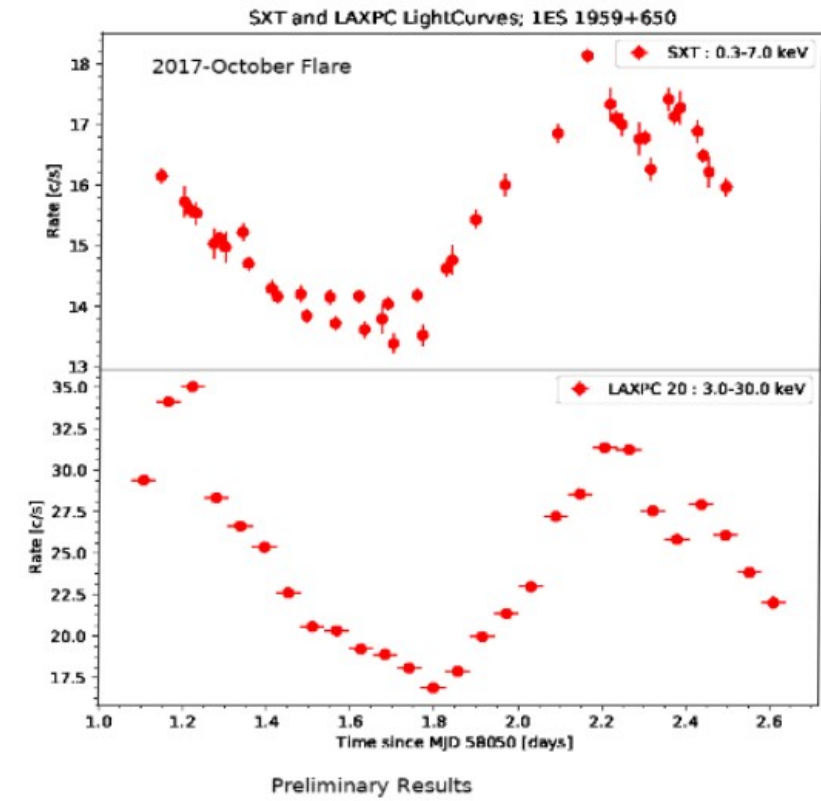
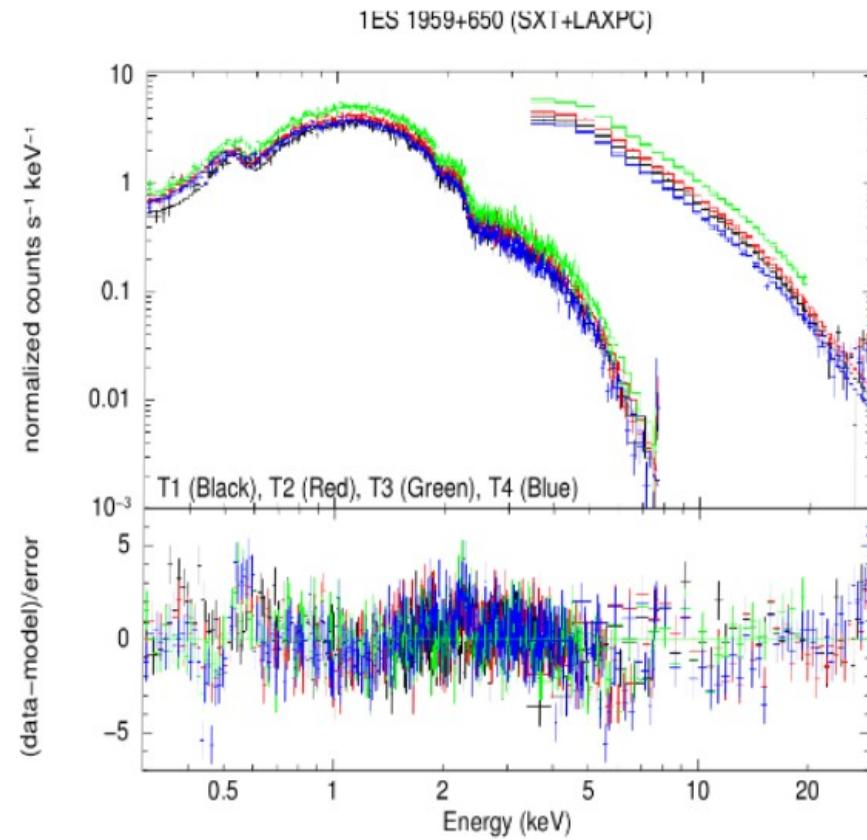
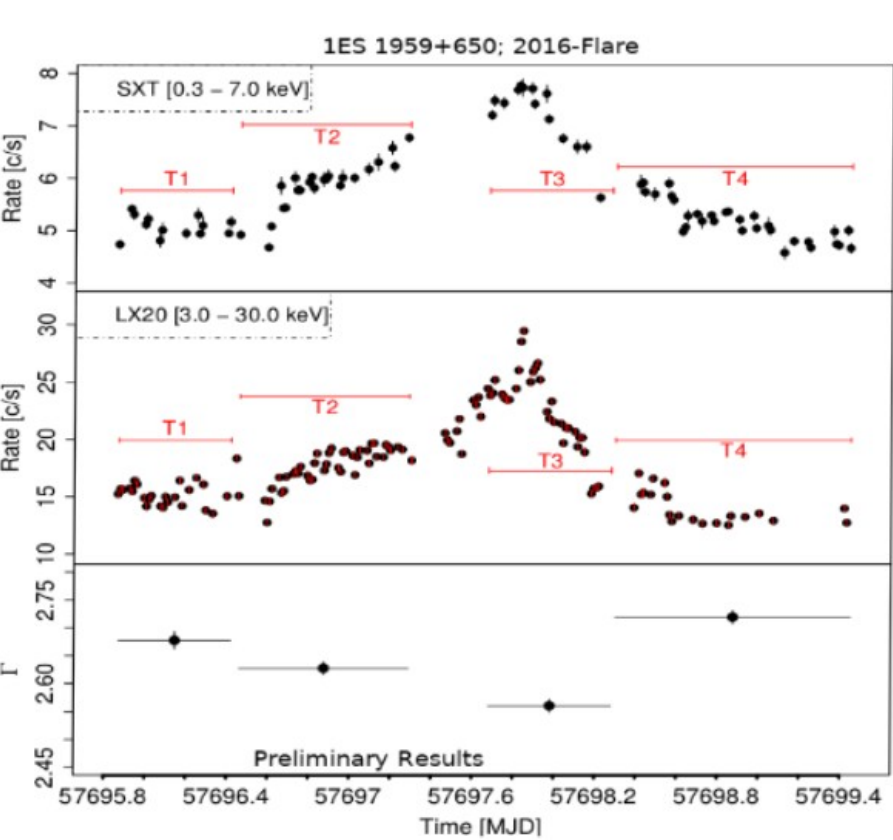


FIG. 7.— Power density spectra at three different energy bands 3.0–8.0 keV (first column), 8.0–20.0 keV (second column) and 20.0–80.0 keV (third column) are shown for three observations when strong QPOs are detected at ~ 2.55 Hz (top rows), ~ 4.53 Hz (middle rows) and ~ 6.55 Hz (bottom rows) respectively. Due to observed break in noise continuum, broken powerlaw model is used to fit the noise component while Lorentzians are used to fit QPO and harmonic features. It may be noted that a significant, excess noise component (modeled with broad Lorentzian) appears in the PDS with higher QPO frequencies at the energy > 8.0 keV. Such features were not detected by *RXTE* due to its highly reduced efficiency in 8.0–20.0 keV energy band.

HBL PKS 2155-304; SXT Lightcurves

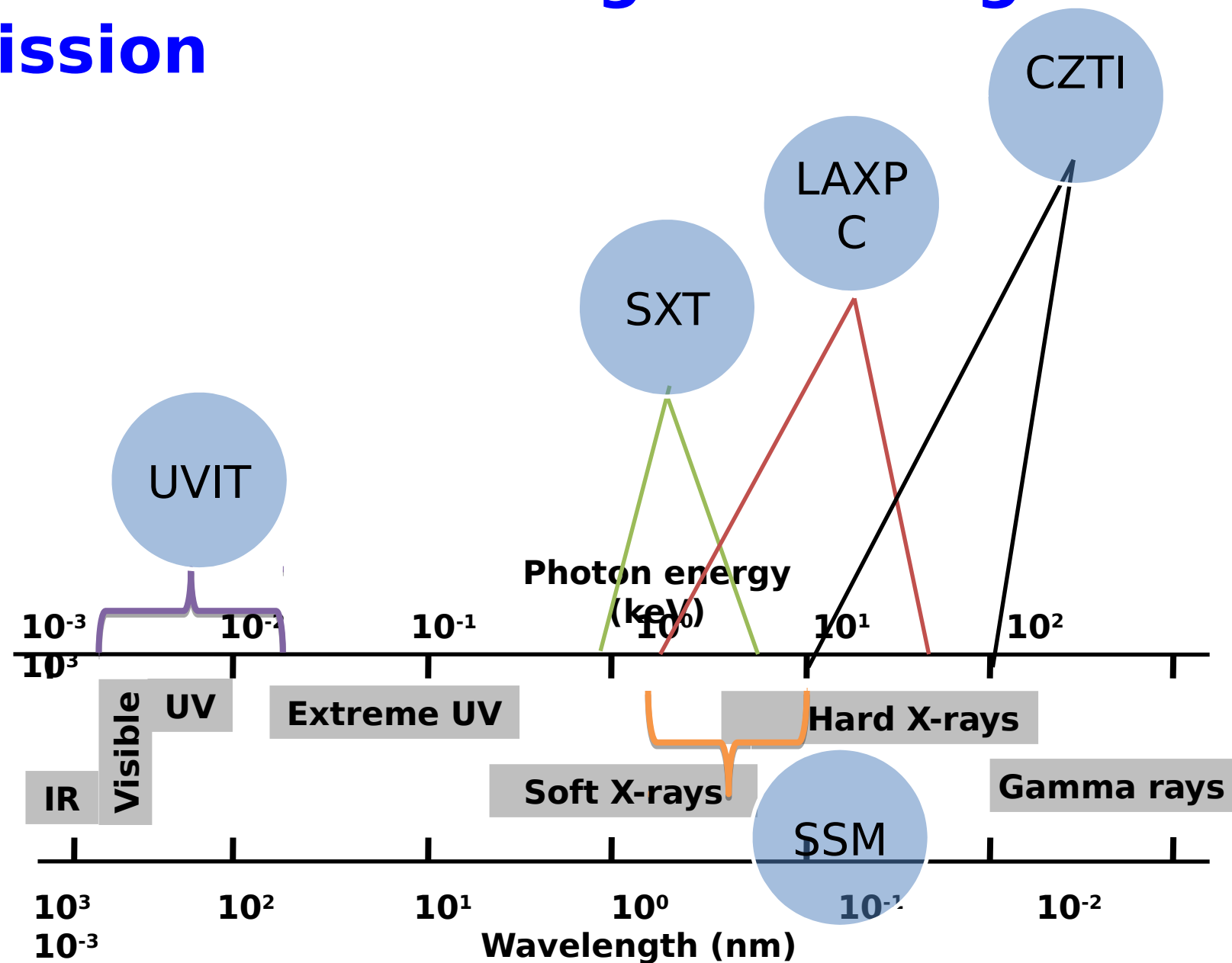






Chandra et al. 2018b, In Prep

Broadband coverage in a single mission

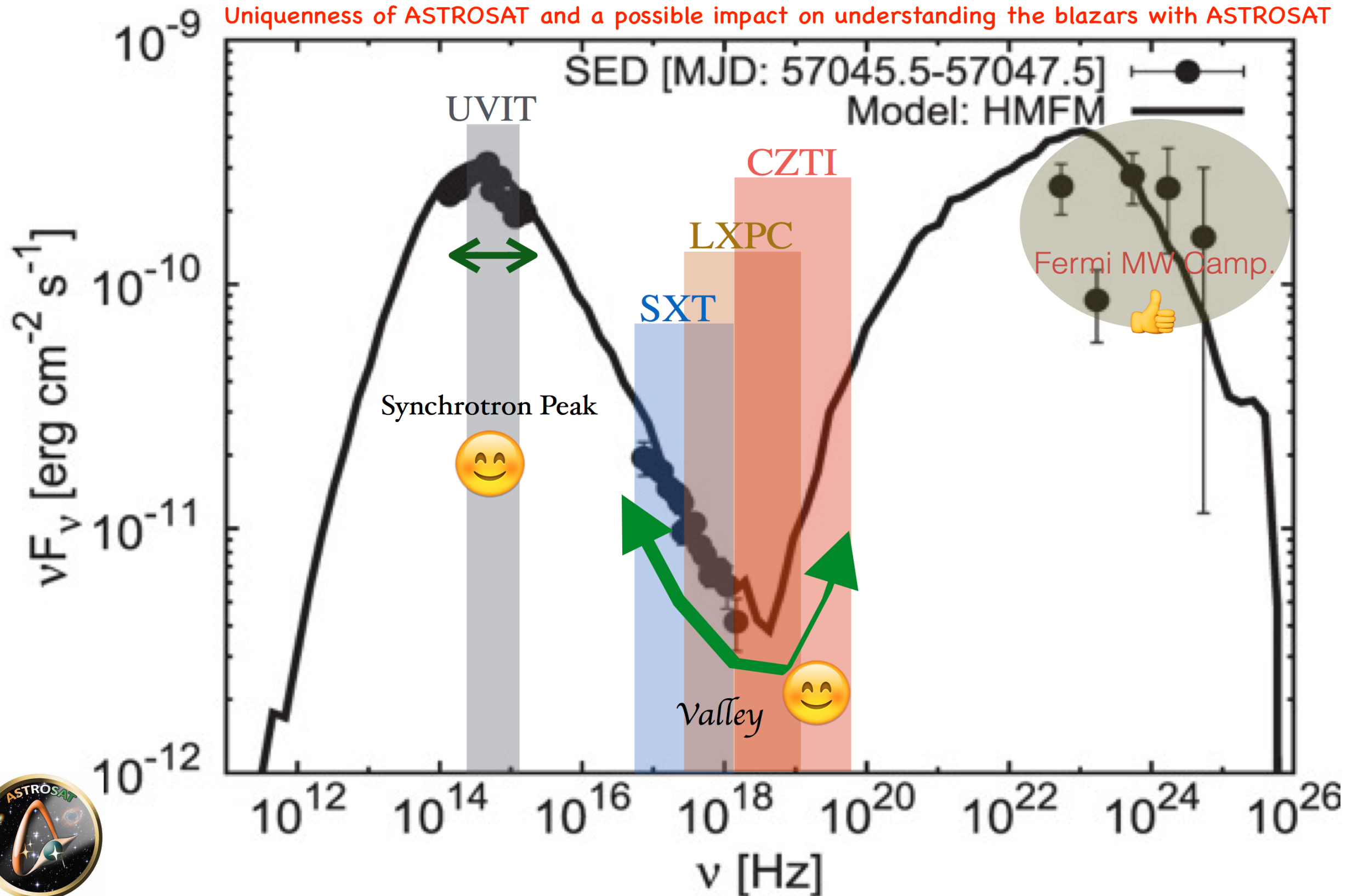


Typical Electromagnetic Spectrum



ASTROSAT AND BLAZARS??

Uniqueness of ASTROSAT and a possible impact on understanding the blazars with ASTROSAT



AstroSat Observations Cycles

0 - 6 months (Oct. 2015 – Mar. 2016): Performance Verification (Complete)

6-12 months (Apr. 2016 – Sep. 2016): Guaranteed Time Observations (GTO) (Complete)

Year 2 (Oct. 2016 – Sep. 2017): AO: Indian PIs: 35% and GTO: 50% (complete)

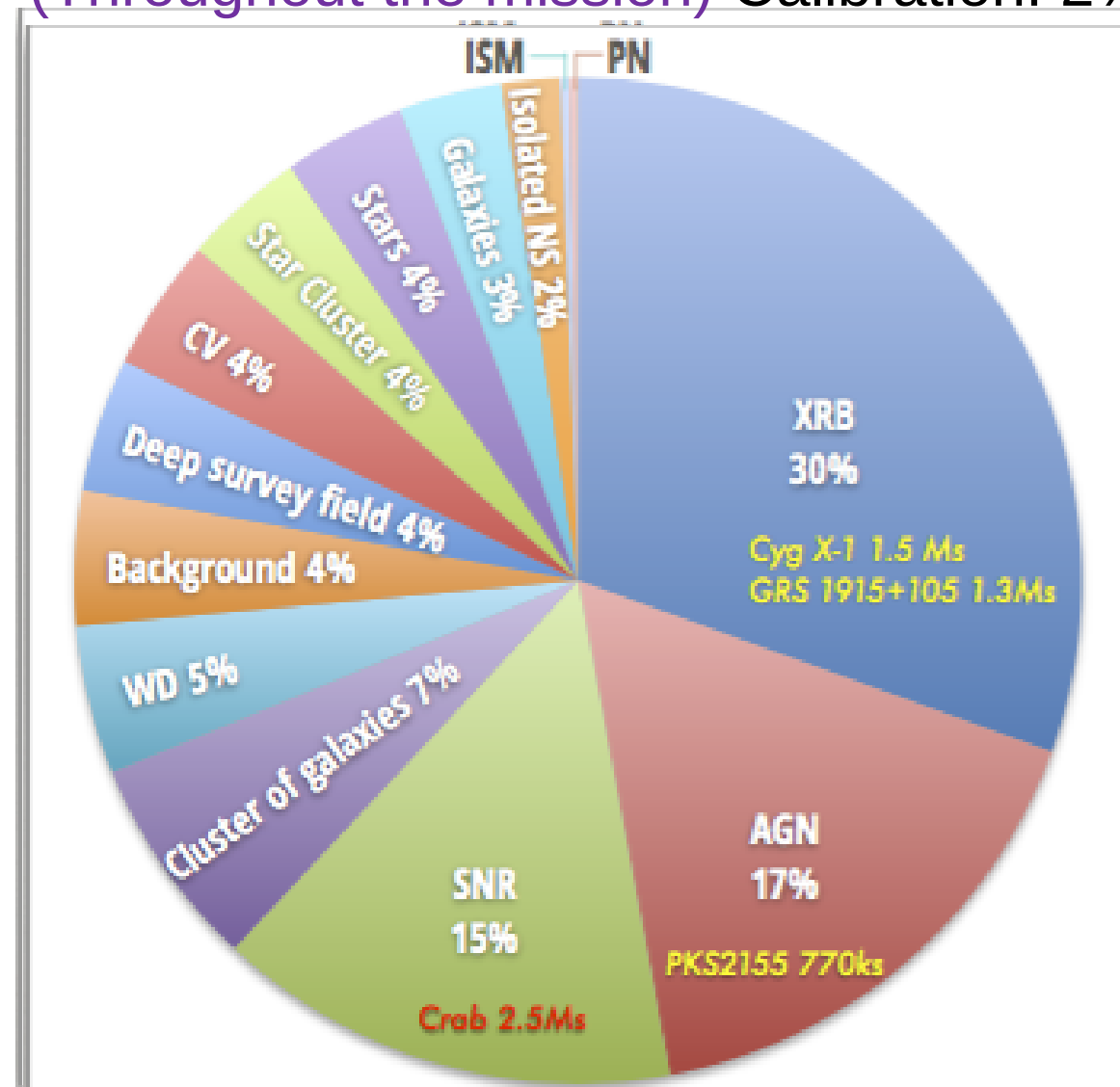
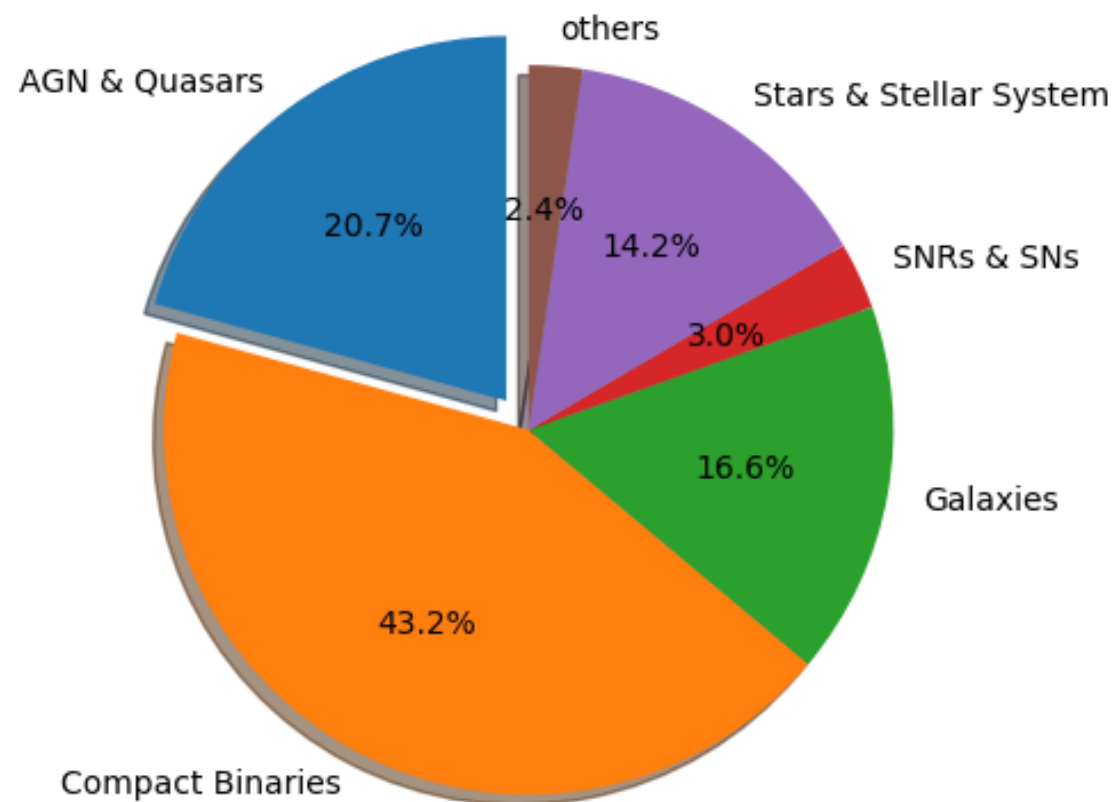
Year 3 (Oct. 2017 – Sep. 2018): AO: Indian PIs: 45%, International PIs: 10% and GTO: 30% (Ongoing....)

Year 4 (Oct. 2018 – Sep. 2019): AO: Indian PIs: 65%, International PIs: 20% **(Proposal selection don, observations starting next month)**

Year 5 (Oct. 2019 – Sept 2020) AO : Indian PIs: 65%, International PIs: 20%

From Year 2: CSA: 5%, Leicester: 3%,
Target of Opportunity (ToO): 5%
(Throughout the mission) Calibration: 2%

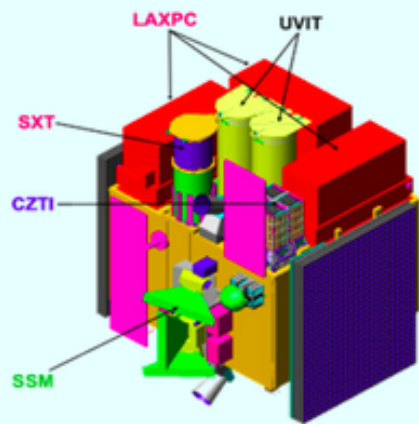
AstroSat Proposals Accepted for A04 & G08 Cycle:
01 October 2017 - 30 September 2018



1st year : 140 sources, 337 targets

www.tifr.res.in/~astrosat_sxt

ASTROSAT



ASTROSAT is the first dedicated Indian astronomy satellite. It was launched from Sriharikota, India on September 28, 2015. It is a multi-wavelength (Optical + UV + X-ray) astronomy space mission in a near-Earth, near-equatorial orbit. For more detail click : [HERE](#).

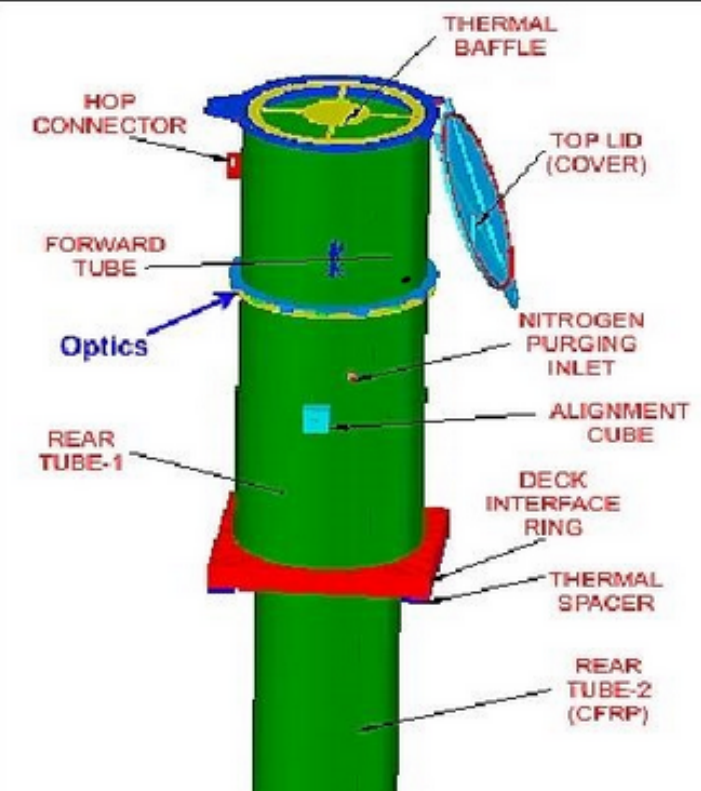
News

- 📅 SXT Level2 pipeline version 1.4a released on 06 Dec 2017
- 📅 SXT Camera Door Opened (First Light Seen) on 26 Oct 2015
- 📅 SXT Telescope Door Opened on 15 Oct 2015
- 📅 SXT switched on 30 Sep 2015
- 📅 SXT aboard ASTROSAT launched on 28 Sep 2015

X-RAY TELESCOPE

Telescopes make clear images of cosmic objects by collecting large amount of light using mirrors and by focusing them on a detector at the focal point. In case of X-rays, the reflections from the mirrors have to be at very small (grazing) angles. So the geometry of an X-ray telescope is quite different from that of a telescope which works in visible light.

One such geometry is the Wolter-I geometry, in which X-rays are reflected twice, first by a paraboloid mirror and then by a hyperboloid mirror, before being focused. Usually many nested mirror shells are used to increase the collecting area. The surface quality of the mirror is required to be very good (roughness of a few Angstroms) to avoid scattering, and therein lies a challenge.





ASTROSAT SXT

Prelim. SXT Data Quality Check Report

Last updated on: 2018-09-14T10:54:20

Created by: Sunil Chandra

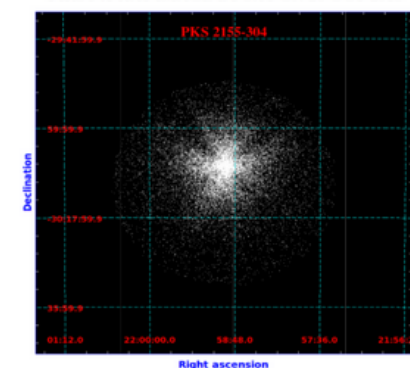
Maintained by: Nilima Kamble

Important Links : a) SXT POC TIFR

b) IUCAA Science Support Cell

c) ISRO AstroSat

FIRST X-RAY IMAGE BY SXT ON 26-OCT-2015



Data Folder	OBSID	Observer	Object	RA	Dec	Exposure [s]	Mode	Date/Time Start	MJD Start	MJD Stop
Search...										
AS1A02_005T01_9000000948sxtPC00_level2	A02_005T01_9000000948	ritaban	MRK421	166.114	38.209	110692.33	PC	2017-01-03T13:31:35	57756.6	57761.4
AS1A02_006T01_9000000776sxtPC00_level2	A02_006T01_9000000776	gulabd	NGC1365	53.402	-36.14	22645.6	PC	2016-11-07T11:11:30	57699.5	57700.4
AS1A02_006T01_9000000802sxtPC00_level2	A02_006T01_9000000802	gulabd	NGC1365	53.402	-36.14	20065.87	PC	2016-11-17T04:40:28	57709.2	57709.9
AS1A02_006T01_9000000934sxtPC00_level2	A02_006T01_9000000934	gulabd	NGC1365	53.402	-36.14	22612.55	PC	2016-12-27T08:32:32	57749.4	57750.3
AS1A02_024T01_9000001044sxtPC00_level2	A02_024T01_9000001044	labani	NGC4388	186.445	12.662	52177.97	PC	2017-02-19T22:32:33	57803.9	57806.0
AS1A02_027T01_9000001084sxtPC00_level2	A02_027T01_9000001084	dleahy	HERX-1	254.458	35.342	25462.73	PC	2017-03-15T15:39:50	57827.7	57829.0
AS1A02_028T01_9000000724sxtPC00_level2	A02_028T01_9000000724	dleahy	M31NO.1	10.711	41.25	16333.24	PC	2016-10-10T07:06:5	57671.3	57671.8
AS1A02_028T03_9000000788sxtPC00_level2	A02_028T03_9000000788	dleahy	M31NO.2	11.037	41.557	22902.35	PC	2016-11-11T14:51:55	57703.6	57704.5
AS1A02_029T01_9000001058sxtPC00_level2	A02_029T01_9000001058	sudip	4U1724-30	261.889	-30.802	223.48	PC	2017-02-27T07:00:30	57811.3	57811.3
AS1A02_029T01_9000001082sxtPC00_level2	A02_029T01_9000001082	sudip	4U1724-30	261.889	-30.802	1635.66	PC	2017-03-14T22:17:16	57826.9	57827.0
AS1A02_029T01_9000001386sxtFW0E_level2	A02_029T01_9000001386	sudip	4U1724-30	261.889	-30.802	12175.42	FW	2017-07-16T00:30:46	57950.0	57950.6
AS1A02_029T01_9000001416sxtPC00_level2	A02_029T01_9000001416	sudip	4U1724-30	261.889	-30.802	216.35	PC	2017-07-29T15:31:34	57963.7	57963.7
AS1A02_031T01_9000001014sxtPC00_level2	A02_031T01_9000001014	sarita	NGC5053	199.113	17.7	8839.39	PC	2017-02-08T17:31:33	57792.8	57793.1
AS1A02_046T01_9000000814sxtPC00_level2	A02_046T01_9000000814	rita	HOLMBERG-IIX-I	124.54	70.698	11842.14	PC	2016-11-21T10:05:38	57713.4	57713.8
AS1A02_046T01_9000000864sxtPC00_level2	A02_046T01_9000000864	rita	HOLMBERG-IIX-I	124.54	70.698	13026.17	PC	2016-12-08T16:09:32	57730.7	57731.4
AS1A02_055T01_9000001114sxtPC00_level2	A02_055T01_9000001114	veeresh	B31702+457	255.877	45.68	82170.49	PC	2017-03-25T05:43:26	57837.2	57840.7
AS1A02_058T02_9000001030sxtPC00_level2	A02_058T02_9000001030	omkar	NGC4262	184.877	14.878	6713.98	PC	2017-02-14T15:39:2	57798.7	57798.8
AS1A02_058T03_9000000856sxtPC00_level2	A02_058T03_9000000856	omkar	NGC1533	62.466	-56.118	3592.36	PC	2016-12-07T13:20:18	57729.6	57729.6
AS1A02_058T03_9000001870sxtPC00_level2	A02_058T03_9000001870	omkar	NGC1533	62.466	-56.118	4060.72	PC	2018-01-24T18:16:42	58142.8	58143.0
AS1A02_063T01_9000000702sxtFW0E_level2	A02_063T01_9000000702	sriramou	CYGX-2	326.171	38.321	12957.97	FW	2016-10-02T22:31:25	57664.0	57664.5
AS1A02_070T01_9000000832sxtPC00_level2	A02_070T01_9000000832	savithri	MRK0926	346.181	-8.686	21663.8	PC	2016-11-26T22:32:40	57718.9	57720.3
AS1A02_071T01_9000001078sxtPC00_level2	A02_071T01_9000001078	sudip	GX349+2	256.435	-36.423	309.07	PC	2017-03-12T14:05:20	57824.6	57824.6
AS1A02_072T02_9000000892sxtFW0E_level2	A02_072T02_9000000892	omiti	PGC26406	175.805	30.005	15220.76	FW	2017-01-25T22:30:12	57770.0	57770.7

Summary : [Observation Details](#) | [Lightcurve + Image](#) | [Attitude + Spectrum](#) |

- exposure: 6183.80701277 second
- sourceid: PKS2155-304
- observer: DANIELA
- ra_pnt: 329.716937
- dec_pnt: -30.225589

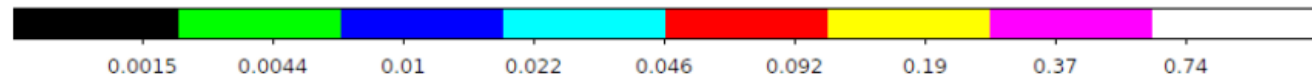
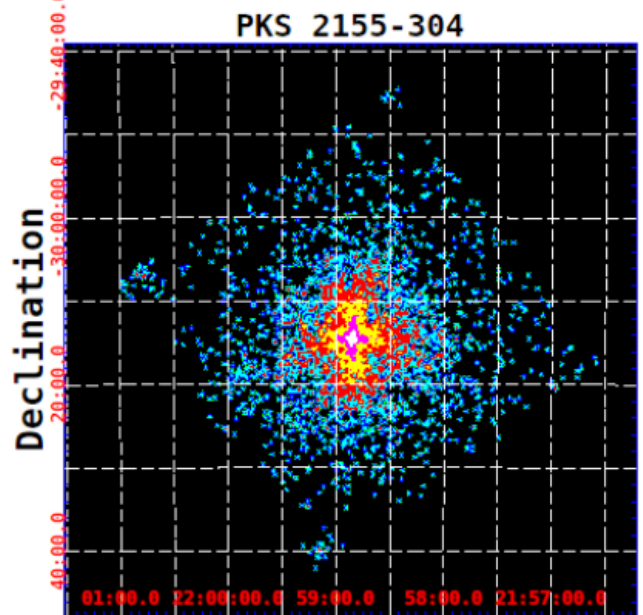
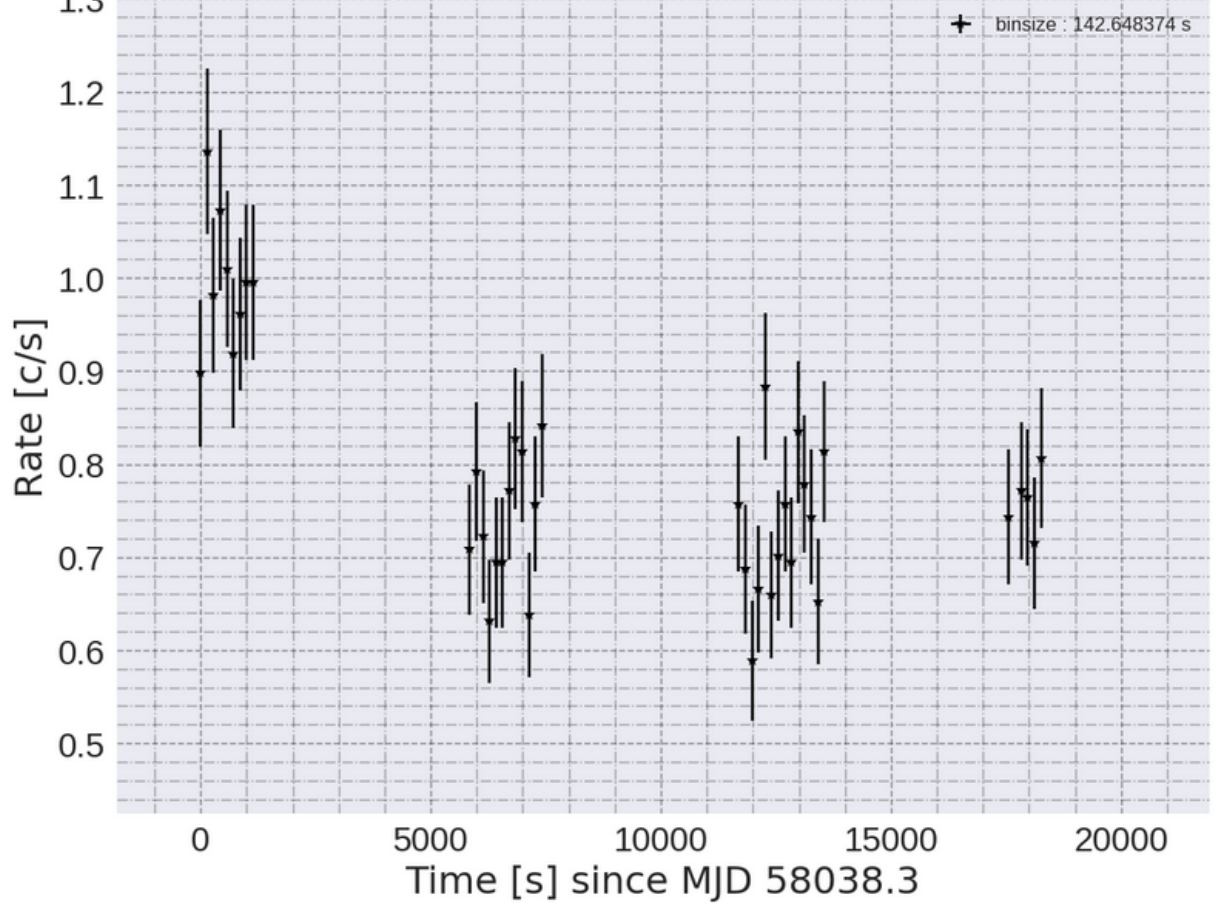
Lightcurve (count rate v/s time) and Image

^TOP

Full Frame Lightcurve

Sky Image

SXT Lightcurve; Object : PKS2155-304; E: 0.3-7.0 keV; Exp. : 6.2 ks



58038.3 58038.34 58038.37 58038.4 58038.44 58038.47 58038.5 58038.54

SUMMARY

- ▶ ASTROSAT provides a unique facility with unprecedented energy coverage over the EM spectrum
- ▶ All the instruments on-board are now working in optimized mode
- ▶ The results from all the instruments are very much encouraging
- ▶ The simultaneous broad-energy energy ... ideal instrument for correlation studies ...connections between Synchrotron, SSC and EC
- ▶ The MW campaigns ... expected to provide us unique asset to improve the understanding of physical processes in a range of astrophysical systems
- ▶ The international community is having a dedicated block for observing time

THANK YOU



Sunil Chandra